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## Highlights

- Up-to-date reported test and numerical results of chord sidewall failure in RHS joints were collated.
- Effects of brace-to-chord height ratio, brace angle, steel grade and chord stress ratio were evaluated.
- Two design methods were proposed for chord sidewall failure in RHS joints under brace axial compression.
- Design of chord sidewall failure in RHS joints under brace axial tension and brace bending was discussed.

# Design of chord sidewall failure in RHS joints using steel grades up to S960

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**Abstract:** It is well known that the current design rules adopted by international design codes such as ISO 14346 and design guides, e.g., the CIDECT design guide No. 3, for chord sidewall failure in mild steel RHS joints under brace axial compression are considerably conservative, if the RHS joints are adequately supported out-of-plane. This paper presents an investigation into chord sidewall failure in rectangular hollow section (RHS) joints using steel grades up to S960. Representative existing design methods for chord sidewall failure in RHS joints are reviewed, and two alternative design methods, i.e., the modified bearing-buckling method and the Lan-Kuhn method, are proposed. Up-to-date test and numerical results reported in the literature are compiled. A wide range of geometric parameters, steel grades up to S960 and loading cases of brace axial loading, brace in-plane bending and brace out-of-plane bending are covered. The existing and proposed design methods are assessed against the collated results. The effects of brace-to-chord height ratio, brace angle, steel grade and chord stress ratio are evaluated. It is shown that the proposed design methods can provide more consistent resistance predictions for chord sidewall failure in mild steel and high-strength steel RHS joints under brace axial compression. Corresponding user-friendly design rules are suggested. The design of chord sidewall failure in RHS joints under brace axial tension, brace in-plane bending and brace out-of-plane bending is discussed. Further required research on, in particular, high-strength steel RHS joints is highlighted.

**Keywords:** Rectangular hollow section; X joints; T joints; Y joints; Chord sidewall failure; Design rules

## 1. Introduction

Rectangular hollow sections (RHS) exhibit an aesthetic appearance and feature excellent structural efficiency especially with regard to loading of compression and torsion because of the closed shape. The evident advantages of RHS result in wide applications in structural, mechanical, transport and offshore fields. The connection of RHS members is vitally crucial for the structural integrity, and direct welding of the intersecting brace to the through chord is the simplest and cleanest solution for the connection. Design rules are needed for such welded RHS joints to facilitate structural applications.

Fig. 1 shows the configurations and notations of RHS-to-RHS X, T and Y joints. Chord sidewall failure is a typical failure mode in full-width RHS joints with brace-to-chord width ratio ( $\beta$ ) of 1.0. In the 1970s, test data for chord sidewall failure in RHS X joints became available from Czechowski and Brodka [1] and Barentse [2]. Czechowski

41 and Brodka [1] developed an empirical equation based on their data which showed a large scatter. This is probably  
42 because in the Polish tests the brace and chord of the X joints were fabricated from cold-formed channel sections  
43 with fabrication tolerances. Furthermore, distortion of the chord cross-section resulted in a sway-type failure mode  
44 because of the pinned-end support at the brace ends and the inadequate out-of-plane support at the chord ends.  
45 Barentse [2] assessed various local buckling models against his test results. Brodka and Szlendak [3] and Kato and  
46 Nishiyama [4] proposed analytical models which appear to be too complicated for design. Later, Wardenier and  
47 Davies [5] developed a simpler combined bearing-buckling model based on a conservative lower bound of the  
48 aforementioned Polish and Dutch test results. It adopts a combined check for the bearing resistance using the steel  
49 yield stress of the chord ( $f_{y0}$ ) and the local buckling capacity employing a local buckling stress of the chord sidewall  
50 ( $f_k$ ). The value of  $f_k$  can be determined using the relevant Eurocode buckling curves [6] or equivalent buckling  
51 curves. The bearing-buckling method is adopted by various design codes, e.g., EN 1993-1-8 [7] and ISO 14346  
52 [8], and design guides such as the CIDECT design guide No. 3 [9-10] and the IIW recommendations [11-13] for  
53 chord sidewall failure.

54  
55 Extensive research on chord sidewall failure has been conducted since the mid-1980s. Davies et al. [14]  
56 summarised various design methods and the influence of different joint parameters for chord sidewall failure in  
57 RHS X joints. Packer [15] conducted tests on 31 full-width RHS X joints to supplement the existing test database  
58 of 40 RHS X joints reported in the literature, and concluded that the codified bearing-buckling method for chord  
59 sidewall failure was too conservative. Giddings and Wardenier [16] compiled CIDECT Monograph No. 6 in which  
60 various state-of-the-art theories at that time for chord sidewall failure were summarised. Davies and Roodbaraky  
61 [17] examined the effect of brace angle ( $\theta_1$ ) on the resistances of various failure modes in RHS X joints using the  
62 results of tests as well as elastic and elastic-plastic numerical analyses reported by Platt [18]. It was found that the  
63 average resistance enhancement for decreasing brace angles could be quantified by the brace angle function of  
64  $(1/\sin\theta_1)^{0.5}$  for chord sidewall failure in RHS X joints under brace axial compression and tension. Yu [19] proposed  
65 a four-hinge yield line model and assumed that the chord sidewall was fully clamped for chord sidewall failure in  
66 RHS-to-RHS X and T joints subjected to brace axial compression, brace in-plane bending and brace out-of-plane  
67 bending. Becque and Cheng [20] conservatively assumed that the chord sidewall is pinned along the chord length  
68 direction, and proposed a plate buckling model to predict the buckling initiation of the chord sidewall in RHS-to-  
69 RHS X joints. Kuhn et al. [21] proposed an equation of the buckling reduction factor which is linearized against  
70 chord height to wall thickness ratio ( $2\gamma^*=h_0/t_0$ ) for chord sidewall failure in RHS X joints under brace axial  
71 compression. This simplifies the determination of  $f_k$  values without using the buckling curves. Wardenier [22]  
72 proposed to modify the codified resistance equation to consider the effect of brace-to-chord height ratio ( $\eta^*=h_1/h_0$ ).  
73 Lan et al. [23-24] developed an analytical model for plate buckling to properly consider the beneficial restraint of  
74 the chord face and brace for the chord sidewall in RHS-to-RHS X and T joints. Comprehensive assessment of the  
75 design methods remains limited, and more suitable design rules for chord sidewall failure in RHS joints are needed.

76  
77 This study aims to evaluate existing design methods and to propose suitable design methods and design rules for  
78 chord sidewall failure in mild steel and high-strength steel RHS joints. Test and numerical results of RHS X and  
79 T joints reported in the literature have been collated. A wide range of geometric parameters, steel grades up to  
80 S960 and loading cases of brace axial loading, brace in-plane bending and brace out-of-plane bending were

81 covered. Existing design methods and proposed design methods in this study were assessed against the compiled  
82 results. The effects of brace-to-chord height ratio, brace angle, steel grade and chord stress were evaluated. Design  
83 rules were proposed for chord sidewall failure in mild steel and high-strength steel RHS joints under brace axial  
84 compression. The design of chord sidewall failure in RHS joints under brace axial tension, brace in-plane bending  
85 and brace out-of-plane bending was discussed. Further research on, in particular, high-strength steel RHS joints  
86 was highlighted.

87

## 88 **2. Design methods for chord sidewall failure**

89

### 90 **2.1. General**

91

92 This section elaborates the representative design methods in the literature and the proposed design methods in this  
93 study for chord sidewall failure in RHS joints. The bearing-buckling model proposed by Wardenier and Davies [5]  
94 is widely adopted by international design codes, e.g., EN 1993-1-8 [7] and ISO 14346 [8], and design guides such  
95 as the CIDECT design guide No. 3 [9-10] and the IIW recommendations [11-13]. The design rules specified in  
96 these design codes and design guides are nearly the same for chord sidewall failure in mild steel RHS joints. Kuhn  
97 et al. [21] and Wardenier [22], among others, proposed modifications to the codified design method in order to  
98 reduce the conservatism and scatter of the resistance predictions. Other analytical models were also proposed for  
99 chord sidewall failure, e.g., the four-hinge yield line model in combination with a reduced chord sidewall buckling  
100 length proposed by Yu [19], and the plate buckling models proposed by Becque and Cheng [20] and Lan et al.  
101 [23-24]. Two alternative design methods, i.e., a modified bearing-buckling method and a so-called “Lan-Kuhn  
102 method” are proposed herein. These design proposals are summarised in the subsequent sections. It is noted that  
103 steel with a grade up to S355 is defined as mild steel in this study.

104

### 105 **2.2. Codified bearing-buckling model**

106

107 Fig. 2 shows the codified bearing-buckling model for chord sidewall failure in RHS joints under brace axial loading  
108 [7, 8, 25]. It is based on a combined check for the bearing resistance using the steel yield stress of the chord ( $f_{y0}$ )  
109 and the local buckling capacity employing the local buckling stress of the chord sidewall ( $f_k$ ). The  $f_k$  values can be  
110 obtained using the relevant Eurocode buckling curves [6] or equivalent buckling curves. Chord sidewall failure is  
111 conservatively considered as the buckling of a pinned-end strut with a buckling length of  $h_0-2t_0$ . The spreading of  
112 the normal component of the brace load ( $N_1 \sin \theta_1$ ) is assumed to be over a length of  $h_1/\sin \theta_1 + 5t_0$  at each chord  
113 sidewall with a dispersion slope of 2.5 to 1 through the chord thickness. This results in the following basic  
114 resistance equation for chord sidewall failure in mild steel RHS joints under brace axial loading:

$$115 \quad N_{1,Rd} = \frac{f_k t_0}{\sin \theta_1} \left( \frac{2h_1}{\sin \theta_1} + 10t_0 \right) Q_f \quad (1)$$

116 where  $t_0$  is the chord sidewall thickness,  $h_1$  is the brace height,  $\theta_1$  is the brace angle (see Fig. 2) and  $Q_f$  is a chord  
117 stress function which accounts for the effect of longitudinal chord stresses. The term  $f_k$ , which equals  $f_{y0}$  for brace  
axial tension, is the buckling stress of the chord sidewall for brace axial compression, and is taken as [7, 8, 25]:

$$f_k = \begin{cases} 0.8\chi_C f_{y0} \sin \theta_1 & \text{for X joints} \\ \chi_C f_{y0} & \text{for T and Y joints} \end{cases} \quad (2)$$

118 where  $\chi_C$  is a buckling reduction factor for column buckling according to EN 1993-1-1 [6], or a comparable design  
119 code, for a normalized slenderness ( $\lambda_C$ ) defined by [7, 8, 25]:

$$\lambda_C = \frac{3.46 \left( \frac{h_0}{t_0} - 2 \right) \sqrt{\frac{1}{\sin \theta_1}}}{\pi \sqrt{\frac{E}{f_{y0}}}} \quad (3)$$

120 The Eurocode buckling reduction factor ( $\chi_C$ ) can be obtained from tables as a function of the normalized  
121 slenderness or by substituting Eq. (3) into Eqs. (4-5) where  $\alpha$  is an imperfection factor. For cold-formed steel cross-  
122 sections, a buckling curve c with  $\alpha=0.49$  is used, and a buckling curve a with  $\alpha=0.21$  is adopted for hot-finished  
123 steel cross-sections using steel grades up to S420.

$$\chi = \frac{1}{\varphi + \sqrt{\varphi^2 - \lambda^2}} \leq 1.0 \quad (4)$$

$$\varphi = 0.5(1 + \alpha(\lambda - 0.2) + \lambda^2) \quad (5)$$

124  
125 It is noted that the  $f_k$  value is reduced by including  $\sin \theta_1$  in the  $f_k$  function for X joints (see Eq. (2)) and by  
126 incorporating the term of  $(1/\sin \theta_1)^{0.5}$  in the  $\lambda_C$  equation (see Eq. (3)). This is because the research conducted by  
127 Platt [18] showed that the effect of  $\theta_1$  on the resistance of a chord sidewall, in an RHS X joint with  $\theta_1 < 90^\circ$ , is  
128 considerably smaller than being proportional to  $1/\sin \theta_1$ . Furthermore, a reduction factor of 0.8 (see Eq. (2)) (i.e., a  
129 safety factor of 1.25) was adopted for RHS X joints to increase the safety margin for the X joints with higher chord  
130 sidewall slenderness ( $h_0/t_0$ ) which exhibit less-ductile failure.

131  
132 Initially, no chord stress function ( $Q_f$ ) was included for chord sidewall failure because the influence of small chord  
133 stresses is insignificant. Later on, based on the research by Wardenier et al. [26], the following  $Q_f$  functions, which  
134 are the same for  $\beta=1.0$ , were adopted for RHS T, Y and X joints [8, 25]:

$$Q_f = (1 - |n|)^{0.6-0.5\beta} \quad \text{for chord compression stress } (n < 0) \quad (6)$$

$$Q_f = (1 - |n|)^{0.1} \quad \text{for chord tension stress } (n \geq 0) \quad (7)$$

135 where  $n$  is the normal (longitudinal) stress ratio in the chord connecting face. The  $n$  value is taken as the sum of  
136 the ratio of the chord axial force ( $N_{0,Ed}$ ) to the chord axial yield capacity ( $N_{pl,0,Rd}$ ) and the ratio of the chord bending  
137 moment ( $M_{0,Ed}$ ) to the chord plastic moment capacity ( $M_{pl,0,Rd}$ ). Negative and positive  $n$  values denote chord  
138 compression and tension stresses, respectively.

139  
140 The resistance equations for chord sidewall failure in mild steel RHS X, T and Y joints under brace axial loading  
141 have been extended for brace in-plane bending (see Eq. (8)) and for brace out-of-plane bending (see Eq. (9)) as  
142 follows [8, 10, 25 ]:

$$M_{ip,1,Rd} = 0.5\chi_C f_{y0} t_0 (h_1 + 5t_0)^2 Q_f \quad (8)$$

$$M_{op,1,Rd} = \chi_C f_{y0} t_0 (b_0 - t_0) (h_1 + 5t_0) Q_f \quad (9)$$

143 which are conservative for  $\theta_1 < 90^\circ$ . For brace out-of-plane bending, it is presumed that the chord distortion failure  
144 mode is prevented.

145

146 The aforementioned developments resulted in the resistance equations summarized in Table 1, which have been  
147 adopted by recent international design codes and design guides [8, 10, 13, 25]. Up to 2013, the design  
148 recommendations applied to a nominal yield stress ( $f_{y0}$ ) of the finished hollow section up to 460 MPa, with the  $f_{y0}$   
149 value for design not exceeding 0.8 times the ultimate stress of the chord ( $f_{u0}$ ). The stipulated joint resistances in  
150 the design recommendations [8, 10, 13] were to be multiplied by a material factor ( $C_f$ ) of 0.90 for  $355 \text{ MPa} < f_{y0} \leq$   
151  $460 \text{ MPa}$ . The most recent prEN 1993-1-8 [25] has proposed:  $C_f = 1.00$  for  $f_{y0} \leq 355 \text{ MPa}$ ,  $C_f = 0.90$  for  $355 \text{ MPa}$   
152  $< f_{y0} \leq 460 \text{ MPa}$ ,  $C_f = 0.86$  for  $460 \text{ MPa} < f_{y0} \leq 550 \text{ MPa}$ , and  $C_f = 0.80$  for  $550 \text{ MPa} < f_{y0} \leq 700 \text{ MPa}$ .

153

### 154 2.3. Modifications to codified bearing-buckling model

155

#### 156 2.3.1. Linearized buckling reduction factor proposed by Kuhn et al. [21]

157

158 Kuhn et al. [21] showed that the column buckling reduction factor ( $\chi_{0.5}$ ) for mild steel cold-formed RHS decreases  
159 in an approximately linear manner with increasing  $h_0/t_0$  ratio up to 50. The  $\chi_{0.5}$  value was obtained using a reduced  
160 chord sidewall slenderness ( $\lambda_{0.5}$ ), which was first suggested by Yu [19]:

$$\lambda_{0.5} = 0.5 \lambda_c \quad (10)$$

161 It is assumed that the chord sidewall is fixed along the longitudinal edges, and thus the  $\lambda_{0.5}$  value is taken as half  
162 of that adopted in Table 1. Kuhn et al. [21] proposed to express the buckling reduction factor as a linear function  
163 of the  $h_0/t_0$  ratio and also to include empirical terms of  $(1/\sin\theta_1)^{0.5}$  and  $(f_{y0}/350)^{0.5}$  to consider the effects of brace  
164 angle and steel grade. These proposals resulted in the following linearized equation of the buckling reduction factor  
165 for RHS X joints having  $h_1/(h_0 \sin\theta_1) > 0.25$  [21]:

$$\chi_{Kuhn} = 1.15 - 0.013 \frac{h_0}{t_0} \sqrt{\frac{1}{\sin\theta_1}} \sqrt{\frac{f_{y0}}{350}} \leq 1.0 \quad (11)$$

166 For plate-to-RHS X joints and RHS-to-RHS X joints with  $h_1/(h_0 \sin\theta_1) \leq 0.25$ ,  $\chi_{Kuhn} = 1.0$  is proposed to be used  
167 within the general validity range given in Table 1, and the resistance for chord sidewall failure in RHS X joints  
168 under brace axial compression can be obtained from [21]:

$$N_{Kuhn} = \chi_{Kuhn} f_{y0} t_0 \left( \frac{2h_1}{\sin\theta_1} + 10t_0 \right) Q_f \quad (12)$$

169 It is noted that the term of  $f_k t_0 / \sin\theta_1$  in Eq. (1) becomes  $\chi f_{y0} t_0$  when substituting  $f_k = \chi f_{y0} \sin\theta_1$  for RHS X joints.

170

171 The moment capacities for chord sidewall failure in RHS X joints under brace in-plane bending ( $M_{ip,Kuhn}$ ) and  
172 brace out-of-plane bending ( $M_{op,Kuhn}$ ) may be obtained from Eqs. (8-9), but replacing  $\chi_C$  with  $\chi_{Kuhn}$  in Eq. (11).

173

#### 174 2.3.2. $\eta^*$ correction proposed by Wardenier [22]

175

176 Apart from using the reduced chord sidewall slenderness ( $\lambda_{0.5}$ ) for RHS joints that are sufficiently restrained  
 177 against out-of-plane movements, Wardenier [22] proposed to reconsider the effect of brace-to-chord height ratio  
 178 ( $\eta^*=h_1/h_0$ ). This is because the numerical results of Yu [19] and Lan et al. [23-24] show that full-width RHS X  
 179 and T joints with higher  $\eta^*$  and  $2\gamma^*$  ( $=h_0/t_0$ ) ratios have a more-abrupt chord sidewall failure mode, i.e., the load-  
 180 deformation curve exhibits a sharp drop in load after the peak load. Thus, it would be logical to increase at least  
 181 the safety margin for RHS joints with a less-ductile failure mode.

182

183 Wardenier [22] proposed to include a correction function of  $(h_1/h_0)^{-0.15}$  in the resistance equation (see Eq. (1)) in  
 184 order to increase the safety margin for full-width RHS joints with a less-ductile failure mode. The modified  
 185 resistance equations for RHS X joints with  $\theta_1=90^\circ$  and under brace axial compression then become:

$$N_{\text{Ward}} = f_{k,\text{Ward}} t_0 (2h_1 + 10t_0) \left( \frac{h_0}{h_1} \right)^{0.15} Q_f \quad (13)$$

$$f_{k,\text{Ward}} = \chi_{\text{Ward}} f_{y0} \quad (14)$$

186 where  $\chi_{\text{Ward}}$  is the buckling reduction factor obtained using the Eurocode buckling curve c and the chord sidewall  
 187 slenderness ( $\lambda_{0.5}$ ) or the linearized approximation, e.g., as proposed by Kuhn et al. [21] (see Eq. (11)). Using Eq.  
 188 (13) would result in an equal or higher safety margin for the less-ductile RHS joints when compared with the more-  
 189 ductile joints with low  $\eta^*$  and  $2\gamma^*$  ratios.

190

191 The moment capacities for chord sidewall failure in RHS X joints under brace in-plane bending and brace out-of-  
 192 plane bending may be obtained using Eqs. (8-9), but replacing  $\chi_C$  with  $\chi_{\text{Ward}}$ . It is also worth noting that the initial  
 193 analyses conducted by Wardenier [22] indicate that the brace angle effect needs to be reconsidered.

194

## 195 **2.4. Representative analytical models**

196

### 197 *2.4.1. Four-hinge yield line model proposed by Yu [19]*

198

199 In the 1990s, Yu [19] conducted an extensive study on uniplanar and multiplanar RHS joints. A four-hinge yield  
 200 line model (see Fig. 3) was proposed for chord sidewall failure in RHS-to-RHS X and T joints under brace axial  
 201 compression, brace in-plane bending and brace out-of-plane bending. The corresponding resistance equation for  
 202 mild steel RHS-to-RHS X and T joints, with  $\theta_1=90^\circ$  and under brace axial compression, is as follows:

$$N_{\text{Yu}} = 4\chi_{0.5} (\sqrt{\gamma} + \eta) f_{y0} t_0^2 \quad (15)$$

203 where  $\gamma$  ( $=b_0/2t_0$ ) is the chord width to twice chord wall thickness ratio,  $\eta$  ( $=h_1/b_0$ ) is the brace height to chord  
 204 width ratio, and  $\chi_{0.5}$  is the buckling reduction factor determined by substituting  $\lambda_{0.5}$  (see Eq. (10)) into Eqs. (4-5).

205 The four-hinge yield line model assumes that the chord sidewalls are fixed along the longitudinal edges.

206

207 The moment capacity of chord sidewall failure in mild steel RHS-to-RHS T and X joints, with  $\theta_1=90^\circ$  and loaded  
 208 under brace in-plane bending, is given by:



$$M_{ip,Yu} = \chi_{ip,0.5} \left( 2\sqrt{\gamma} + \gamma\eta + \frac{1}{2\eta} \right) f_{y0} t_0^2 h_1 \quad (16)$$

209 where  $\chi_{ip,0.5}$  is the buckling reduction factor which equals 1.0 for  $\eta \leq 1$  and for  $1 < \eta \leq 2$ , is determined by:

$$\chi_{ip,0.5} = 1 + (\eta - 1) \left( \frac{1}{\varphi + \sqrt{\varphi^2 - \lambda_{0.5}^2}} - 1 \right) \quad (17)$$

210 The moment capacity of chord sidewall failure in mild steel RHS-to-RHS T and X joints, with  $\theta_1 = 90^\circ$  and loaded  
211 under brace out-of-plane bending, is given by:

$$M_{op,Yu} = \chi_{0.5} \left( \sqrt{2(1+2\gamma)} + 2\gamma\eta \right) f_{y0} t_0^2 b_1 \quad (18)$$

212

#### 213 2.4.2. Plate buckling model proposed by Becque and Cheng [20]

214

215 Becque and Cheng [20] proposed a plate buckling model conservatively assuming that the chord sidewall is pinned  
216 along the longitudinal edges for chord sidewall failure in RHS-to-RHS X joints under brace axial compression.

217 The corresponding resistance equation is as follows:

$$N_{Becque} = 2.4 \chi_{Becque} f_{y0} h_1 t_0 \quad (19)$$

218 where  $\chi_{Becque}$  is the buckling reduction factor obtained using the relevant buckling curve, e.g., according to Eqs.  
219 (4-5); however, a modified imperfection factor  $\alpha = 0.08$  is recommended and the proposed chord sidewall  
220 slenderness is as follows:

$$\lambda_{Becque} = \sqrt{\frac{P_y}{P_{cr}}} = \sqrt{\frac{2.4 f_{y0} h_1 t_0}{2 f_{cr,Becque} h_1 t_0}} \quad (20)$$

$$f_{cr,Becque} = 1.346 \frac{\pi^2 E}{12(1-\nu^2)} \frac{t_0^2}{h_0 h_1} \quad (21)$$

221 where  $E$  is the steel elastic modulus and  $\nu$  is the Poisson ratio taken as 0.3.

222

223 It is noted that this design method is proposed to predict the initiation of buckling of the chord sidewall. This  
224 buckling load can be considerably lower than the joint resistance determined by the peak load or the load at an  
225 indentation limit of  $3\%b_0$ , whichever occurs at a smaller deformation, which is commonly adopted in other studies.  
226 This design method is therefore not included in the subsequent evaluation.

227

#### 228 2.4.3. Plate buckling model proposed by Lan et al. [23-24]

229

230 The restraint from the chord face and the brace to the chord sidewall is stronger than that of a pinned-end boundary  
231 condition, but weaker than that of fixed edges. Lan et al. [23] proposed an analytical model of plate buckling for  
232 chord sidewall failure in RHS-to-RHS X joints which can properly consider the restraint and utilize the strain  
233 hardening of steel materials by using the continuous strength method. Later, Lan et al. [24] simplified the resistance  
234 equations without considering the strain hardening for RHS-to-RHS X and T joints to reduce the computational  
235 effort. It is noted that the strain hardening in high-strength steel is not pronounced. Fig. 4 shows the proposed plate  
236 buckling model for RHS-to-RHS X and T joints with  $\theta_1 = 90^\circ$ .

237

238 The elastic buckling stress equation proposed for RHS-to-RHS T and X joints with  $\theta_1=90^\circ$ , which can properly  
239 consider the restraint from the chord face and the brace to the chord sidewall, is as follows [23-24]:

$$f_{cr,Lan} = 3.2 \frac{\pi^2 E}{12(1-\nu^2)} \left( \frac{t_0}{h_0} \right)^{1.96} \left( \frac{h_0}{h_1} \right)^{0.66} \quad (22)$$

240 The overall cross-section slenderness of the chord sidewall is defined by [23-24]:

$$\lambda_{Lan} = \sqrt{\frac{f_{y0}}{f_{cr,Lan}}} \quad (23)$$

241 which can be obtained by substituting Eq. (22),  $E=210000$  MPa and  $\nu=0.3$  into Eq. (23):

$$\lambda_{Lan} = 0.024 \left( \frac{h_0}{t_0} \right)^{0.98} \left( \frac{h_1}{h_0} \right)^{0.33} \sqrt{\frac{f_{y0}}{355}} \quad (24)$$

242 which can be conservatively approximated by rounding off the exponents:

$$\lambda_{Lan} = 0.024 \frac{h_0}{t_0} \left( \frac{h_1}{h_0} \right)^{0.3} \sqrt{\frac{f_{y0}}{355}} \quad (25)$$

243

244 The plate buckling reduction factor ( $\chi_{Lan}$ ) neglecting the strain hardening, which is based on the base curves  
245 proposed by Lan et al. [23], is as follows [24]:

$$\chi_{Lan} = \begin{cases} 1.0 & \text{for } \lambda_{Lan} \leq 0.6 \\ 0.8 \left( 1 - \frac{0.2}{\lambda_{Lan}^{1.6}} \right) \frac{1}{\lambda_{Lan}^{1.6}} & \text{for } \lambda_{Lan} > 0.6 \end{cases} \quad (26)$$

246 The curve of the  $\chi_{Lan}$  equation is relatively linear for  $\chi_{Lan}>0.6$  and is herein suggested to be approximated by:

$$\chi_{Lan} = 1.39 - 0.67\lambda_{Lan} \leq 1.0 \quad (27)$$

247 which can be obtained by substituting Eq. (25) into Eq. (27) for steel grades up to S960 and ratios of  $b_0/t_0$  and  $h_0/t_0$   
248 up to 40:

$$\chi_{Lan} = 1.39 - 0.016 \frac{h_0}{t_0} \left( \frac{h_1}{h_0} \right)^{0.3} \sqrt{\frac{f_{y0}}{355}} \leq 1.0 \quad (28)$$

249 The linearized buckling reduction factor (see Eq. (28)) can produce conservative resistance prediction for RHS  
250 joints using higher steel grades in combination with ratios of  $b_0/t_0$  and  $h_0/t_0$  larger than 35, and thus the original  
251 Eqs. (25-26) are suggested for such cases.

252

253 The joint resistance ( $N_{Lan}$ ) for chord sidewall failure in RHS-to-RHS T and X joints with  $\theta_1=90^\circ$  can be obtained  
254 from [24]:

$$N_{Lan} = \chi_{Lan} f_{y0} t_0 (2h_1 + 10t_0) Q_f \quad (29)$$

255 The joint resistance for chord sidewall failure in RHS-to-RHS T, Y and X joints under brace in-plane bending and  
256 brace out-of-plane bending may be obtained from Eqs. (8-9), but replacing  $\chi_C$  with  $\chi_{Lan}$  in Eq. (28). The linearized  
257 Lan method using Eqs. (28-29) will be examined in the subsequent analyses.

258

## 259 2.5. Proposed design methods

260

### 261 2.5.1. General

262

263 Lan et al. [23-24, 27-28] evaluated the material effect on the resistance of fabricated RHS and CHS X and T joints  
264 under brace axial compression and proposed the following equation for the material factor ( $C_f$ ) to quantify the  
265 resistance reduction, which was resulted from the material effect:

$$C_f = 1.1 - 62 f_{y0} / E \leq 1.0 \quad (30)$$

266 An equivalent  $C_f$  equation as a function of only  $f_{y0}$  is proposed in this study to maintain a uniform format for  
267 equations:

$$C_f = 1.1 - 0.1 f_{y0} / 355 \leq 1.0 \quad (31)$$

268 The differences between the calculated  $C_f$  values using Eqs. (30-31) are found to be marginal. The derived  $C_f$   
269 values are 1.00, 0.97, 0.90, 0.85 and 0.83 for steel grades of S355, S460, S700, S900 and S960, respectively. The  
270 corresponding rounded-off  $C_f$  values of 1.00, 0.95, 0.90, 0.85 and 0.80 may be used for chord sidewall failure  
271 under brace compression loading, which are more optimistic than the general  $C_f$  values stipulated in prEN 1993-  
272 1-8 [25]. Eq. (31) is incorporated in the proposed design methods mainly because significant material softening in  
273 the heat-affected zone of high-strength steel can occur in practice and the effect of fabrication imperfections can  
274 be more pronounced for chord sidewall failure in high-strength steel RHS joints (see Section 5).

275

276 The codified bearing-buckling method adopts various compensations for the brace angle effect in RHS X joints  
277 by including  $\sin\theta_1$  in  $f_k$  and  $(1/\sin\theta_1)^{0.5}$  in  $\lambda_c$  (see Section 2.2). It is noted that the correction of  $\sin\theta_1$  and safety  
278 factor of 0.8 in  $f_k$  are not adopted for RHS T and Y joints (see Eq. (2)). This leads to inconsistencies for the design  
279 of RHS X and T/Y joints. The brace angle effect for RHS X, T and Y joints is herein recommended to be  
280 approximated by only using a function of  $(1/\sin\theta_1)^{0.5}$  in the final resistance equation, which is in line with Davies  
281 and Roodbaraky [17].

282

283 Wardenier [22] proposed to adopt a correction function of  $(h_1/h_0)^{-0.15}$  for the design joint resistance (see Eq. (13)).  
284 However, it is more suitable, e.g., for the loading case of brace axial tension, to include the term of  $(h_1/h_0)^{-0.15}$  in  
285 the  $f_k$  function and to impose an upper limit of  $f_{y0}$  for  $f_k$  values, and thus the design joint resistance can be limited  
286 by the yield resistance.

287

288 The aforementioned proposed modifications result in the basic resistance equation for chord sidewall failure in  
289 RHS X, T and Y joints under brace axial compression as follows:

$$N_p = C_f f_{k,p} t_0 (2h_1 + 10t_0) \sqrt{\frac{1}{\sin\theta_1}} Q_f \quad (32)$$

$$f_{k,p} = \chi_p \left( \frac{h_0}{h_1} \right)^{0.15} f_{y0} \leq f_{y0} \quad (33)$$

290 where  $C_f$  is the proposed material factor (see Eq. (31)),  $t_0(2h_1+10t_0)$  is the bearing area taken in line with the  
 291 bearing-buckling model,  $Q_f$  is the chord stress function (see Eqs. (6-7)),  $(1/\sin\theta_1)^{0.5}$  is the brace angle function,  $f_{k,P}$   
 292 is the buckling stress of the chord sidewall and  $\chi_P$  is the proposed buckling reduction factor. Two alternative  
 293 methods are proposed to derive the  $\chi_P$  values in this study.

294

### 295 2.5.2. Modified bearing-buckling method

296

297 Some code committees prefer, as currently used, a design method for chord sidewall failure which adopts the  
 298 column buckling curve, in order to maintain consistency between the design of RHS joints and that of members.  
 299 A format similar to the current set-up in the design codes and design guides was thus employed. The proposed  
 300 modified bearing-buckling method in this study adopts a buckling curve c with  $\alpha=0.49$  in EN 1993-1-1 [6] and a  
 301 reduced chord sidewall slenderness of  $\lambda_{0.5}$  (see Eq. (10)). The buckling reduction factor ( $\chi_{P,M}$ ) can therefore be  
 302 obtained from:

$$\chi_{P,M} = \frac{1}{\varphi_M + \sqrt{\varphi_M^2 - \lambda_{0.5}^2}} \leq 1.0 \quad (34)$$

$$\varphi_M = 0.5(1 + 0.49(\lambda_{0.5} - 0.2) + \lambda_{0.5}^2) \quad (35)$$

303 where  $\chi_{P,M}$  is the proposed modified buckling reduction factor. It is noted that the codified bearing-buckling method  
 304 and the four-hinge yield line model adopt different buckling curves according to the fabrication methods of cross-  
 305 sections (e.g., cold-formed or hot-finished). However, a buckling curve c is herein suggested for all cross-sections  
 306 to simplify the design process and to produce resistance predictions on the conservative side.

307

### 308 2.5.3. Lan-Kuhn method

309

310 The linearized Kuhn method is based on the combined bearing-buckling model with the chord sidewall assumed  
 311 to be fixed along the longitudinal edges and local buckling covered by the strut buckling coefficient [5,29], whereas  
 312 the linearized Lan method is based on a plate local buckling model. In reality, bearing governs for low  $h_0/t_0$  ratios  
 313 and local buckling dominates for higher  $h_0/t_0$  ratios. Therefore, the Lan-Kuhn method using a linearized function  
 314 of buckling reduction factor is proposed in this study.

315

316 The effect of  $h_1/h_0$  ratio is not considered in the Kuhn method. The Lan method adopts a term of  $(h_1/h_0)^{0.3}$  in the  
 317 buckling reduction factor (see Eq. (28)) to quantify the effect, and this approach was initially considered for the  
 318 Lan-Kuhn method. However, it was found that this could result in large deviations of the predicted resistances  
 319 especially for  $\eta^* \neq 1.0$  when compared with the proposed modified bearing-buckling method. It is noted that two  
 320 alternative design methods should give comparable resistances. More detailed discussions can be found in  
 321 Wardenier et al. [30]. Therefore, the influence of  $h_1/h_0$  ratio is included in the buckling stress equation (see Eq.  
 322 (33)), and the effect of  $\theta_1$  is considered in the basic resistance equation (see Eq. (32)) in this study. Only the effects  
 323 of the  $h_0/t_0$  ratio and  $f_{y0}$  are quantified in the proposed equation for the buckling reduction factor ( $\chi_{P,LK}$ ):

$$\chi_{P,LK} = 1.12 - 0.012 \frac{h_0}{t_0} \sqrt{\frac{f_{y0}}{355}} \leq 1.0 \quad (36)$$

324

325 The joint resistance for chord sidewall failure in RHS X, T and Y joints under brace in-plane bending and brace  
326 out-of-plane bending may be obtained from Eqs. (8-9), but replacing  $\chi_C f_{y0}$  with  $f_{k,P}$  in Eq. (33).

327

#### 328 2.5.4. Comparison of the buckling reduction factors

329

330 Table 2 shows a comparison of  $\chi_{P,LK}$ ,  $\chi_{Kuhn}$  and  $\chi_{Lan}$  with  $\chi_{P,M}$  for  $h_0/t_0 \leq 40$ ,  $h_1/h_0 = 1.0$ ,  $235 \text{ MPa} \leq f_{y0} \leq 960 \text{ MPa}$  and  
331  $\theta_1 = 90^\circ$ . The  $\chi_{P,LK}/\chi_{P,M}$ ,  $\chi_{Kuhn}/\chi_{P,M}$ , and  $\chi_{Lan}/\chi_{P,M}$  ratios equal the corresponding resistance ratios because  $h_1/h_0 = 1.0$   
332 and the same basic resistance equation (Eq. (1)) is adopted. It is shown that the differences between  $\chi_{P,LK}$  and  $\chi_{P,M}$   
333 are minor with a maximum deviation of 4%. These two proposed design methods therefore give almost equivalent  
334 resistances. The  $\chi_{Kuhn}$  equation also produces excellent approximations of  $\chi_{P,M}$  values for lower steel grades;  
335 however, it is observed that the  $\chi_{Kuhn}$  value deviates from the  $\chi_{P,M}$  value for steel grades of S700 and higher in  
336 combination with a high  $h_0/t_0$  ratio. The maximum deviation is 4% for steel grades up to S700 and becomes 16%  
337 for S960 which is on the conservative side. It is also shown that  $\chi_{Lan}$  values are generally higher than  $\chi_{P,M}$  values  
338 with a maximum discrepancy of 20% because the Lan method is based on a plate buckling model and is not related  
339 to the column buckling curves. It should be noted that the deviations of  $\chi_{Kuhn}$  and  $\chi_{Lan}$  values from  $\chi_{P,M}$  values could  
340 be larger for  $h_1/h_0 < 1.0$  and  $h_1/h_0 > 1.0$  because the effect of the  $h_1/h_0$  ratio is not considered in  $\chi_{Kuhn}$  for the Kuhn  
341 method; however, it is considered in  $\chi_{Lan}$  for the Lan method and in  $f_{k,P}$  for the proposed modified bearing-buckling  
342 method. More detailed information can be found in Wardenier et al. [30].

343

### 344 3. Evaluation of design methods for full-width RHS X and T joints under brace axial compression

345

#### 346 3.1. General

347

348 A database of test and numerical results totalling 248 full-width RHS X joints under brace axial compression  
349 reported in the literature was established. Results of plate-to-RHS X joints were analysed by Kuhn et al. [21] and  
350 are not further considered in this study. The compiled results were adopted to evaluate the following six design  
351 methods:

- 352 (1) The bearing-buckling method, but using the Eurocode buckling curve c and  $\lambda_{0.5}$  with  $N_{C,M}$  defined by Eqs. (1)  
353 and (10)
- 354 (2) The Kuhn linearized method in Section 2.3.1 with  $N_{Kuhn}$  defined by Eqs. (11-12)
- 355 (3) The Yu four-hinge yield line method in Section 2.4.1 with  $N_{Yu}$  defined by Eqs. (10) and (15)
- 356 (4) The Lan plate buckling method using the linearized approach in Section 2.4.3 with  $N_{Lan}$  defined by Eqs. (28-  
357 29)
- 358 (5) The bearing-buckling method, but using the Eurocode buckling curve c,  $\lambda_{0.5}$  and  $(\eta^*)^{-0.15}$  correction in Section  
359 2.5.2 with  $N_{P,M}$  defined by Eqs. (32-35)
- 360 (6) The Lan-Kuhn method using the linearized approach in Section 2.5.3 with  $N_{P,LK}$  defined by Eqs. (32-33) and  
361 (36)

362 The original equations in Section 2 are used in this study unless specified. The corresponding joint resistances  
363 obtained using the six design methods ( $N_{C,M}$ ,  $N_{Kuhn}$ ,  $N_{Yu}$ ,  $N_{Lan}$ ,  $N_{P,M}$  and  $N_{P,LK}$ ) will be compared with the test and

364 numerical resistances ( $N_{1u}$ ) in the subsequent sections. It should be noted that  $N_{C,M}=N_{P,M}$  for  $\eta^*=1.0$ , and the effect  
 365 of the  $\eta^*$  correction could be evaluated by comparing  $N_{C,M}$  with  $N_{P,M}$  for  $\eta^*<1$  and  $\eta^*>1$ .

366

367 It should be noted that the safety factor of 1.25 for RHS X joints, adopted by the aforementioned design codes and  
 368 design guides, was set to be unity in the assessment of the design methods. The Eurocode buckling curve c was  
 369 conservatively used for all RHS joints, regardless of whether tests or numerical models used hot-finished or cold-  
 370 formed hollow sections. In addition, RHS joints with  $N_{1u}/N_y>1.1$ , where  $N_y$  is the joint yield resistance, were  
 371 excluded from the analyses because such data may not be realistic and could lead to a large scatter for the  
 372 subsequent statistical analyses. The  $N_y$  values for all the design methods in this study is obtained from:

$$N_y = f_{y0}t_0(2h_1 + 10t_0)\sqrt{\frac{1}{\sin\theta_1}} \quad (37)$$

373 where the term  $(1/\sin\theta_1)^{0.5}$  is adopted to consider the brace angle effect, in line with Davies and Roodbaraky [17],  
 374 and the  $Q_f$  function is not incorporated. For this comparison, the omission of  $Q_f$  function is conservative, especially  
 375 for large absolute values of chord stress ratio ( $n$ ), as it leads to lower  $N_{1u}/N_y$  ratios.

376

### 377 3.2. Test results of RHS-to-RHS X joints

378

379 Table 3 summarises the compiled test results totalling 51 full-width RHS-to-RHS X-joints under brace axial  
 380 compression. Source references for most tests are given in Kuhn et al. [21] and Fan [31]. Additional test results of  
 381 high-strength steel RHS-to-RHS X joints reported by Feldmann et al. [32] and Pandey and Young [33] were also  
 382 collated. It is shown that five RHS joints have resistances exceeding  $1.1N_y$  and therefore only the remaining 46  
 383 RHS-to-RHS X joints will be included in the subsequent analyses. The parameter ranges for the screened test  
 384 database were  $\beta=1.0$ ,  $12.6\leq 2\gamma\leq 42.2$ ,  $12.6\leq 2\gamma^*\leq 56.9$ ,  $0.50\leq \eta\leq 2.47$ ,  $0.60\leq \eta^*\leq 1.00$ ,  $-0.87\leq n\leq 0$ ,  $44^\circ\leq \theta_1\leq 90^\circ$  and  $228$   
 385  $\text{MPa}\leq f_{y0}\leq 1080 \text{ MPa}$ . Cold-formed and hot-finished RHS were covered.

386

387 The brace angle effect is re-evaluated against the test results of RHS-to-RHS X joints with varying brace angles  
 388 in this study. Davies et al. [14] and Packer [15] found that the effect of brace angle on the resistance of full-width  
 389 RHS X joints is smaller than being proportional to  $1/\sin\theta_1$ . Davies and Roodbaraky [17] reported that, for brace  
 390 axial compression and tension, the enhancement of resistance for decreasing the brace angle could be more  
 391 accurately quantified by a function of  $(1/\sin\theta_1)^{0.5}$ . Therefore, in the current codified design rules (see Table 1), the  
 392 brace angle effect is, based on the initial investigations by Platt [18], minimised by various compensations in the  
 393 chord sidewall slenderness ( $\lambda_c$ ) and the buckling stress ( $f_k$ ) for the X joints. It is noted that the term of  $f_k t_0/\sin\theta_1$  in  
 394 Eq. (1) becomes  $\chi f_{y0} t_0$  when substituting  $f_k = \chi f_{y0} \sin\theta_1$  for RHS X joints. The following two options are assessed  
 395 against test results of 19 selected RHS-to-RHS X joints with  $\theta_1\leq 90^\circ$ :

396 (1) Using the codified term of  $h_1/\sin\theta_1$  in the final resistance equation for  $N_{C,M}$ ,  $N_{Kuhn}$ ,  $N_{Lan}$ ,  $N_{P,M}$  and  $N_{P,LK}$ , and  
 397 also including a term of  $(1/\sin\theta_1)^{0.5}$  in  $\lambda_{0.5}$  for  $N_{C,M}$  and  $N_{P,M}$  and in  $\chi_{Kuhn}$  for  $N_{Kuhn}$  (see Table 4). Including a  
 398  $1/\sin\theta_1$  term in the final resistance equation for  $N_{Yu}$ .

399 (2) Only incorporating a term of  $(1/\sin\theta_1)^{0.5}$  in the final resistance equation for  $N_{C,M}$ ,  $N_{Kuhn}$ ,  $N_{Yu}$ ,  $N_{Lan}$ ,  $N_{P,M}$ , and  
 400  $N_{P,LK}$  (see Table 5).

401

402 The material factor ( $C_f$ ) was not used for all the statistical analyses summarised in Tables 4-5 because the variation  
403 in yield stresses is small. The mean values of  $N_{1u}/N_{C,M}$ ,  $N_{1u}/N_{Yu}$ ,  $N_{1u}/N_{Kuhn}$ ,  $N_{1u}/N_{Lan}$ ,  $N_{1u}/N_{P,M}$  and  $N_{1u}/N_{P,LK}$  ratios  
404 are 1.11, 1.02, 1.13, 0.89, 1.11 and 1.05, respectively, with corresponding coefficients of variation (CoVs) of 0.103,  
405 0.103, 0.105, 0.101, 0.103 and 0.104 for the first approach (Table 4). However, for the second option, the mean  
406 values of  $N_{1u}/N_{C,M}$ ,  $N_{1u}/N_{Yu}$ ,  $N_{1u}/N_{Kuhn}$ ,  $N_{1u}/N_{Lan}$ ,  $N_{1u}/N_{P,M}$  and  $N_{1u}/N_{P,LK}$  ratios are 1.12, 1.05, 1.13, 0.95, 1.12 and  
407 1.12, respectively, with corresponding CoVs of 0.086, 0.079, 0.087, 0.082, 0.086 and 0.085 (Table 5). It is shown  
408 that the CoV values of the various design methods for each option are close. The mean values for the second option  
409 are slightly higher and the corresponding CoV values are about 20% lower when compared with those employing  
410 the first solution. Therefore, only including a term of  $(1/\sin\theta_1)^{0.5}$  in the final resistance equation, which is simpler  
411 and yields more consistent resistance predictions, is recommended to account for the brace angle effect.

412

413 The chord stress effect was assessed against test results of eight available RHS-to-RHS X joints with  $\theta_1=90^\circ$  and  
414 varying chord stress ratios ( $n$ ) summarised in Table 6. The codified  $Q_f$  function was adopted for all the design  
415 methods in the statistical analyses and  $C_f=1.0$  was used for all the mild steel X joints. The mean values of  $N_{1u}/N_{C,M}$ ,  
416  $N_{1u}/N_{Yu}$ ,  $N_{1u}/N_{Kuhn}$ ,  $N_{1u}/N_{Lan}$ ,  $N_{1u}/N_{P,M}$  and  $N_{1u}/N_{P,LK}$  ratios are 1.21, 1.13, 1.22, 1.03, 1.21 and 1.23, respectively,  
417 with corresponding CoVs of 0.088, 0.088, 0.088, 0.089, 0.088 and 0.088. All the design methods yield almost the  
418 same CoVs because only the chord stress ratio is different for each test series and all other parameters are nearly  
419 the same. It is also observed that the resistance ratios, which generally exceed 1.0, increase with increasing absolute  
420 value of  $n$  ratio because for high  $|n|$  values the  $Q_f$  function adopts a conservative lower bound for the chord stress  
421 effect.

422

423 The material effect was evaluated against the screened database of 46 RHS-to-RHS X joints in Tables 7-8. The  
424 approach of only including a term of  $(1/\sin\theta_1)^{0.5}$  in the final resistance equation was adopted for all design methods  
425 to reveal the best performance of these methods. For the design methods without using the proposed  $C_f$  factor (see  
426 Table 7), the mean values of  $N_{1u}/N_{C,M}$ ,  $N_{1u}/N_{Yu}$ ,  $N_{1u}/N_{Kuhn}$ ,  $N_{1u}/N_{Lan}$ ,  $N_{1u}/N_{P,M}$  and  $N_{1u}/N_{P,LK}$  ratios are 1.13, 1.05,  
427 1.15, 0.97, 1.12 and 1.13, respectively, with corresponding CoVs of 0.098, 0.096, 0.098, 0.097, 0.097 and 0.097.  
428 For the design methods using the  $C_f$  factor (see Table 8), the mean values of  $N_{1u}/N_{C,M}$ ,  $N_{1u}/N_{Yu}$ ,  $N_{1u}/N_{Kuhn}$ ,  $N_{1u}/N_{Lan}$ ,  
429  $N_{1u}/N_{P,M}$  and  $N_{1u}/N_{P,LK}$  ratios are 1.17, 1.10, 1.20, 1.01, 1.17 and 1.17, respectively, with corresponding CoVs of  
430 0.092, 0.086, 0.116, 0.116, 0.091 and 0.095. Thus, including the  $C_f$  factor reduces the CoVs except for the Kuhn  
431 method and the Lan method, which both use the linearized approach. This is mainly because the Kuhn and Lan  
432 methods are conservative for the S960 specimens tested by Pandey and Young [33].

433

434 The design methods without using the  $C_f$  factor often yield unconservative resistance predictions for the test  
435 specimens with yield stresses higher than 900 MPa (see Table 7). In contrast, including the  $C_f$  factor in the design  
436 methods leads to safe resistance predictions for all the high-strength steel test specimens (see Table 8). Thus, the  
437 proposed  $C_f$  factor is suggested to consider the material effect. It is also shown that the Yu method and the proposed  
438 modified bearing-buckling method give the lowest CoVs, and other design methods yield slightly higher CoVs. It  
439 should be noted that the low resistance ratios of the X5-S960 specimen (see Table 7) may be attributed to the  
440 material softening in the heat-affected zones and/or an insufficient weld size, as commented by Feldmann et al.

441 [32]. The test evidence for high-strength steel RHS joints remains limited, and more related test and numerical  
442 investigations are needed to assess the material effect comprehensively.

443

444 Most of the RHS-to-RHS X joints in Tables 7-8 have  $\eta^* \approx 1.0$  and there are only two X joints with small  $\eta^*$  values  
445 of 0.60 and 0.75. Thus, the effect of including the  $\eta^*$  correction in the  $f_k$  function cannot be fully revealed in the  
446 overall statistical analyses and has been checked in Section 3.3 using the numerical data.

447

### 448 3.3. Numerical results of RHS-to-RHS X joints

449

450 Table 9 summarises the collated numerical results totalling 173 RHS-to-RHS X joints with  $\theta_1=90^\circ$  reported by Yu  
451 [19] and Kuhn et al. [21]. It is shown that 42 RHS joints have resistances exceeding  $1.1N_y$  and therefore only the  
452 remaining 131 joints will be used in the analyses. The parameter ranges for the screened numerical database were  
453  $\beta=1.0$ ,  $10 \leq 2\gamma \leq 35$ ,  $10 \leq 2\gamma^* \leq 35$ ,  $0.25 \leq \eta \leq 2.00$ ,  $0.21 \leq \eta^* \leq 2.50$ ,  $-0.80 \leq n \leq 0.75$ ,  $\theta_1=90^\circ$  and  $f_{y0}=355$  and 398 MPa. Cold-  
454 formed and hot-finished RHS are included. It is noted that all the RHS-to-RHS X joints had  $\theta_1=90^\circ$  and thus the  
455 brace angle effect cannot be evaluated. The  $C_f$  values for  $f_{y0}=355$  and 398 MPa are 1.00 and 0.99, respectively,  
456 thus the material effect is insignificant for these X joints. Nevertheless, the  $C_f$  factor was adopted for all the design  
457 methods to allow for direct comparison.

458

459 The effect of the  $\eta^*$  ratio was examined against the numerical results of 22 selected RHS-to-RHS X joints with  
460  $n=0$  and  $0.42 \leq \eta^* \leq 2.50$  (see Table 10). The mean values of  $N_{1u}/N_{C,M}$ ,  $N_{1u}/N_{Yu}$ ,  $N_{1u}/N_{Kuhn}$ ,  $N_{1u}/N_{Lan}$ ,  $N_{1u}/N_{P,M}$  and  
461  $N_{1u}/N_{P,LK}$  ratios are 1.19, 1.11, 1.20, 1.05, 1.20 and 1.20, respectively, with corresponding CoVs of 0.105, 0.080,  
462 0.106, 0.051, 0.064 and 0.059. It is shown that including the term of  $(\eta^*)^{-0.15}$  in the  $f_{k,P}$  (see Eq. (33)) of proposed  
463 design methods can reduce the CoV by about 40% when compared with the bearing-buckling method ( $N_{C,M}$ ) and  
464 the Kuhn method ( $N_{Kuhn}$ ) in which the  $\eta^*$  effect is not considered. Incorporating the term of  $(\eta^*)^{0.3}$  in the buckling  
465 reduction factor (see Eq. (28)) of the Lan method can also significantly reduce the CoV and the improvement is  
466 slightly better than that of the proposed design methods. For the Lan-Kuhn model, including the  $(\eta^*)^{0.3}$  term in  
467  $\chi_{P,LK}$  (see Eq. (36)), as used in the Lan method, instead of using the  $(\eta^*)^{-0.15}$  correction in  $f_{k,P}$  (see Eq. (33)), slightly  
468 increases the CoV for the joints in Table 10 from 0.059 to 0.062, and the deviations of  $N_{P,LK}$  from  $N_{P,M}$  become  
469 larger up to 7%. Thus, including the proposed  $\eta^*$  correction in  $f_{k,P}$  is suggested.

470

471 The chord stress effect was assessed against numerical results of 10 selected RHS-to-RHS X joints with varying  
472  $n$  ratios (see Table 11) reported by Yu [19]. The codified chord stress function ( $Q_f$ ) was adopted for all the design  
473 methods in the statistical analyses. The mean values of  $N_{1u}/N_{C,M}$ ,  $N_{1u}/N_{Yu}$ ,  $N_{1u}/N_{Kuhn}$ ,  $N_{1u}/N_{Lan}$ ,  $N_{1u}/N_{P,M}$  and  
474  $N_{1u}/N_{P,LK}$  ratios are 1.34, 1.25, 1.34, 1.15, 1.34 and 1.34, respectively, with corresponding CoVs of 0.074, 0.063,  
475 0.073, 0.073, 0.074 and 0.064. The Yu and Lan-Kuhn methods yield the lowest CoVs and provide the most  
476 consistent strength predictions. It is also found that the resistance ratios, which all exceed 1.0, increase with  
477 increasing absolute values of the  $n$  ratio because the codified  $Q_f$  function employs a conservative lower bound for  
478 the chord stress effect. It is noted that these conclusions also apply to the numerical data with varying  $n$  ratios  
479 reported by Kuhn et al. [21] (see Table 12). Similar observations were reported by Kim et al. [34] for RHS X joints  
480 with  $\beta$  ratio up to 1.0 and with  $f_{y0}=324$  MPa and 798 MPa, and also by Lan et al. [23] for fabricated RHS X joints



481 with  $f_{y0}=460, 690$  and  $960$  MPa. Thus, the need for new chord stress functions is not apparent, and Eqs. (6-7) can  
482 be adopted.

483

484 Table 12 shows the results of statistical analyses for the evaluation of all the design methods against the screened  
485 numerical database of 131 RHS-to-RHS X joints. The mean values of  $N_{1u}/N_{C,M}$ ,  $N_{1u}/N_{Yu}$ ,  $N_{1u}/N_{Kuhn}$ ,  $N_{1u}/N_{Lan}$ ,  
486  $N_{1u}/N_{P,M}$  and  $N_{1u}/N_{P,LK}$  ratios are 1.24, 1.15, 1.25, 1.10, 1.23 and 1.24, respectively, with corresponding CoVs of  
487 0.102, 0.082, 0.104, 0.061, 0.065 and 0.064. It is demonstrated that the Lan method and the proposed design  
488 methods produce the lowest CoVs and thus most consistent resistance predictions.

489

### 490 **3.4. Summary for RHS-to-RHS X joints**

491

492 The overall statistical analyses for the test database (see Tables 7-8) show that the Yu method gives the lowest  
493 CoVs; however, the differences with other design methods are small. The approach of only incorporating a term  
494 of  $(1/\sin\theta_1)^{0.5}$  in the final resistance equation can more accurately quantify the brace angle effect and is preferred.  
495 Including the  $C_f$  factor reduces the CoVs for all the design methods except for the Kuhn method and the Lan  
496 method. Incorporating the  $C_f$  factor in all the design methods can yield safe resistance predictions for high-strength  
497 steel test specimens and is preferred; however, more experimental and numerical studies on high-strength steel  
498 joints are needed to further confirm the proposed  $C_f$  factor. The codified chord stress function ( $Q_f$ ) is more  
499 conservative for larger absolute values of  $n$  ratio. It is noted that only two X joints had small  $\eta^*$  ratios in the test  
500 database, and thus the evaluation of  $\eta^*$  effect is based on the numerical data.

501

502 The overall statistical analyses for the numerical database (see Table 12) show that the Lan method and the  
503 proposed design methods (i.e., the proposed modified bearing-buckling method and the Lan-Kuhn method)  
504 produce the lowest CoVs. It is demonstrated that the effect of the  $\eta^*$  ratio on the joint resistance is significant.  
505 Including the  $\eta^*$  correction in the buckling reduction factor or the buckling stress equation results in more  
506 consistent resistance predictions. Similar to the analyses for the test database, the codified  $Q_f$  function is observed  
507 to be more conservative for large absolute values of the  $n$  ratio. The  $Q_f$  function can be adopted to consider the  
508 chord stress effect. It is noted that the numerical database only covers mild steel and  $\theta_1=90^\circ$ ; thus, the  
509 corresponding effects of steel material and brace angle for RHS-to-RHS X joints cannot be examined.

510

511 It can be concluded that the proposed Lan-Kuhn method gives good correlations with the test data and excellent  
512 correlations with the numerical results, and is better than the Kuhn and Lan methods. The proposed modified  
513 bearing-buckling method produces nearly equivalent resistance predictions when compared with the proposed  
514 Lan-Kuhn method. Thus, it can be adopted as an alternative design method which is in line with the current design  
515 rules employing column buckling curves to determine the joint resistance. Although the Yu method is also very  
516 accurate, the proposed design methods which give designers more insights into the structural behaviour of RHS  
517 joints are easier to use and thus are recommended for RHS-to-RHS X joints. Figs. 5-6 illustrate the comparison of  
518 the test and numerical resistances with those predicted by the proposed design methods, both using the  $C_f$  factor.

519

### 520 **3.5. RHS X joints with only one RHS brace welded to the chord**

521

522 Table 13 summarises the collated test results totalling 22 RHS X joints with an RHS brace welded to one side of  
523 the chord and with the support of a block, a flat plate or a rigid solid base at the opposite side of the chord. It should  
524 be noted that although these test specimens have the physical appearance of RHS T joints, the load transfer was  
525 comparable to that of an X joint without shear in the chord, and thus these specimens were classified as RHS X  
526 joints in line with ISO 14346 [8]. The experimental database consists of test results reported by Barentse [2] for a  
527 welded flat plate support, plus Zhao [35], Pandey and Young [36] and Fan [31] for a rigid solid base. The smaller  
528 brace width on either chord side was taken as  $h_1$  in Table 13. It is noted that the test results of RHS X joints with  
529 an unwelded block support reported by Poloni [37] were not included. This is because the chord cross-sections  
530 used had large  $h_0/t_0$  or  $b_0/t_0$  ratios of 57, and hence were potentially sensitive to fabrication tolerances and deviations  
531 in the test set-up. The chord wall slenderness is also out of the typical parameter ranges commonly adopted in  
532 practice. The RHS X joints with  $N_{1w}/N_y > 1.1$  were excluded from the statistical analyses.

533

534 For the compiled RHS X joints, Kuhn et al. [21] proposed three conditions of the chord sidewall end-restraint  
535 along the chord length direction and corresponding chord sidewall slenderness as follows:

- 536 (a) Fixed-fixed: member or plate welded to two opposite chord sides, with a chord sidewall slenderness of  $\lambda_{0.5}$ .  
537 (b) Fixed-pinned: member or plate welded to one chord side and unwelded to the opposite chord side, with a  
538 chord sidewall slenderness of  $\lambda_{0.7} = 1.4\lambda_{0.5}$ .  
539 (c) Pinned-pinned: plates or supports unwelded to two opposite chord sides, with a chord sidewall slenderness of  
540  $\lambda_{1.0} = 2\lambda_{0.5}$ .

541 According to this classification, the RHS X joints with an RHS brace welded to one chord side and with a plate  
542 support welded to the opposite chord side, tested by Barentse [2], can be categorized as class a. Table 14 shows  
543 that the mean values of  $N_{1w}/N_{P,M}$  and  $N_{1w}/N_{P,LK}$  ratios are 1.07 and 1.08, respectively, with corresponding CoVs of  
544 0.054 and 0.059. It is demonstrated that the proposed design methods are applicable for these RHS X joints.

545

546 The remaining RHS X joints investigated by Zhao [35], Pandey and Young [36], and Fan [31] using a rigid solid  
547 base can be grouped as class b. Thus, a chord sidewall slenderness of  $1.4\lambda_{0.5}$  and a buckling curve c were used to  
548 derive the buckling reduction factor ( $\chi_{P,M1}$ ) and the joint resistance for the proposed modified bearing-buckling  
549 method. For the proposed Lan-Kuhn method, the buckling reduction factor may be obtained from:

$$\chi_{P,LK1} = 1.12 - 0.017 \frac{h_0}{t_0} \sqrt{\frac{f_{y0}}{355}} \quad \text{with } h_0/t_0 \leq 40(355/f_{y0})^{0.5} \text{ but } \leq 40 \quad (38)$$

550 It is noted that the buckling reduction factor for RHS X joints in class b decreases non-linearly with increasing  
551  $h_0/t_0$  ratio, for high yield stress and large chord sidewall slenderness. Thus, the validity of the approach of using  
552  $1.4\lambda_{0.5}$  and the proposed linearized  $\chi_{P,LK1}$  function of Eq. (38) (which can become considerably conservative) has  
553 to be limited by  $h_0/t_0 \leq 40(355/f_{y0})^{0.5}$  but  $\leq 40$ . The proposed  $h_0/t_0$  limits are 40, 40, 35, 28, 25 and 24 for steel grades  
554 of S235, S355, S460, S700, S900 and S960, respectively. Such limits are comparable to the class 3 limit specified  
555 in the current EN 1993-1-1 [6], therefore the chord cross-section can be alternatively limited to class 3. This leaves  
556 only one RHS X joint for S960, and the results of statistical analyses for the screened test database of class b are  
557 shown in Table 15. The mean values of  $N_{1w}/N_{P,M}$  and  $N_{1w}/N_{P,LK}$  ratios are 1.16 and 1.18, respectively, with  
558 corresponding CoVs of 0.130 and 0.141. It is shown that the proposed design methods provide conservative

559 resistance predictions. “RHS X joints” with members unwelded to two opposite chord sides in class c are not  
560 examined in this study, but the chord sidewall slenderness of  $\lambda_{1,0}$  suggested by Kuhn et al. [21] may be used.

561

### 562 **3.6. RHS T and Y joints**

563

564 Yu [19] conducted numerical simulations on chord sidewall failure in full-width RHS-to-RHS T joints. For the T  
565 joints under brace axial compression, the global chord bending at the chord crown was eliminated by applying  
566 compensating moments at the chord ends (i.e.,  $Q_f=1.0$ ). The resistance of one full-width RHS-to-RHS T joint with  
567  $2\gamma=24$  was 1% higher than that of the comparable x11a specimen (see Table 9), and the same design rules were  
568 proposed to be applied to RHS-to-RHS X and T joints. The aforementioned design recommendations developed  
569 for RHS X joints are thus suggested for RHS T joints, which is also line with the current design codes and design  
570 guides. The approach of only including a term of  $(1/\sin\theta_1)^{0.5}$  in the final resistance equation is suggested for RHS  
571 Y joints to consider the brace angle effect, which gives unified design rules for RHS X, T and Y joints.

572

573 Yu [19] also numerically examined the chord stress effect on RHS-to-RHS X and T joints with varying chord  
574 sidewall slenderness and chord stress ratios. It was also shown that the effect of the bending moment could be  
575 considered by the  $Q_f$  function. The plastic moment resistance ( $M_{pl,0,Rd}$ ) for class 1 or 2 chord cross-sections and the  
576 elastic moment resistance ( $M_{el,0,Rd}$ ) for class 3 chord cross-sections could be adopted to calculate the chord stress  
577 ratio. Such recommendations will be incorporated in the subsequent proposed design rules in Section 7.

578

## 579 **4. Discussion on full-width RHS X and T joints under brace axial tension**

580

581 Test data of full-width RHS X and T joints subjected to brace axial tension are available for mild steel and high-  
582 strength steel in the literature; however, reanalyses of the test results are required. De Koning and Wardenier [38]  
583 summarised the up-to-date test results for mild steel up to 1984 and compared the test resistances with those  
584 obtained from the resistance equations for chord sidewall failure and brace failure given by Wardenier [29], which  
585 are nearly identical to those in the current design codes and design guides. These test results confirm the suitability  
586 of the codified design rules for steel grades up to and including S355.

587

588 Contradictory research findings have been reported for RHS joints in higher steel grades. For example, for S450  
589 RHS-to-RHS X and T joints, Becque and Wilkinson [39] recommended the use of material factors for RHS joints  
590 with non-ductile fracture failure modes. For the full-width X joints, brittle chord corner fracture and brace failure  
591 were observed, both with low deformation capacity. In contrast, Björk and Saastamoinen [40] and Tuominen and  
592 Björk [41] concluded, based on an assessment of the design equations in EN 1993-1-8 [7], that no material factors  
593 are required for RHS-to-RHS X joints using S420 and S460 and the joints could be considered as being ductile.  
594 Feldmann et al. [32] suggested material factors of 1.0, 0.90 and 0.80 for steel grades of S500, S700 and S960,  
595 respectively. It is noted that the analyses conducted by Feldmann et al. [32] are based on a comparison of the test  
596 resistances with the Eurocode design resistances, and no separate statistical analyses per failure mode were  
597 conducted. The failure modes observed in tests sometimes deviated from those predicted by EN 1993-1-8 [7],  
598 which incorporates different safety factors in the design equations for various failure modes.

599

600 It is noted that most of the tests have been carried out for RHS joints with square hollow section (SHS) brace and  
601 chord having the same steel grade and wall thickness. Comparison of the resistance equations for the brace  
602 effective width failure with those for chord sidewall failure in RHS joints with  $\theta_1=90^\circ$  shows that the equations  
603 then become rather similar. Furthermore, the material softening in the brace and chord resulting from welding  
604 could vary and thus may alter the failure location. These factors explain the observed change in failure modes for  
605 higher-strength steel joints.

606

607 Therefore, more detailed analyses of the aforementioned test results are needed for chord sidewall failure in full-  
608 width RHS joints under brace axial tension. The resistance and deformation capacity per failure mode need to be  
609 re-evaluated to ascertain whether, for the proposed design methods, lower resistance factors ( $\phi$ ) or higher safety  
610 factors ( $\gamma_M$ ) have to be applied for RHS joints using higher steel grades. Further, it is important that the steel  
611 materials used for tests are representative of those in production specifications.

612

## 613 5. Discussion on RHS X and T joints under brace in-plane bending

614

615 Table 16 shows the compiled numerical results totalling eight full-width RHS-to-RHS X joints under brace in-  
616 plane bending reported by Yu [19]. The numerical resistances ( $M_{1u,ip}$ ) were compared with the yield resistances  
617 ( $M_{y,ip}$ ) obtained from:

$$M_{y,ip} = 0.5f_{y0}t_0(h_1 + 5t_0)^2 Q_f \quad (39)$$

618 It is shown that all joints, except for the x12ie2 specimen, reach the yield resistance ( $M_{y,ip}$ ), and the resistance  
619 ratios ( $M_{1u,ip}/M_{y,ip}$ ) of all joints exceed 1.1 except for the specimens of x11ie2 and x12ie2.

620

621 The collated numerical results were adopted to evaluate the six design methods described in Section 3.1. The  
622 codified resistance equation (see Eq. (8)) was used. However, the  $\chi_C$  in Eq. (8) was replaced with  $\chi_{0.5}$  for the  
623 modified bearing-buckling method ( $M_{C,M,ip}$ ),  $\chi_{Kuhn}$  (see Eq. (11)) for the Kuhn method ( $M_{Kuhn,ip}$ ),  $\chi_{ip,0.5}$  (see Eq. (17))  
624 for the Yu method ( $M_{Yu,ip}$ ) and  $\chi_{Lan}$  (see Eq. (28)) for the Lan method ( $M_{Lan,ip}$ ). The term of  $\chi_C f_{y0}$  in Eq. (8) was  
625 replaced with  $f_{k,P}$  (see Eq. (33)) for the proposed modified bearing-buckling method ( $M_{P,M,ip}$ ) and the Lan-Kuhn  
626 method ( $M_{P,LK,ip}$ ). The Eurocode buckling curve c was conservatively used for all the RHS joints using hot-finished  
627 hollow sections.

628

629 Table 17 summarises the results of the statistical analyses. It is shown that the Yu method gives lowest CoV of  
630 0.072, and the CoVs of all other design methods are relatively large. However, it would be currently difficult to  
631 draw conclusions with respect to the design methods. This is because the  $M_{1u,ip}/M_{y,ip}$  ratios of most of the RHS-to-  
632 RHS X joints are higher than 1.1. Additionally, for all the design methods, the plastic moment resistance, assuming  
633 that the stress within the bearing length of  $(h_1+5t_0)$  all reaches the yield stress ( $f_{y0}$ ), is used and the local buckling  
634 effect is considered by the buckling reduction factor. This means that the strain at the outer part of the bearing  
635 length would be considerably high, which may result in premature fracture failure for high-strength steel RHS  
636 joints. More tests are needed to evaluate the suitability of these design methods for chord sidewall failure in higher-  
637 strength steel RHS joints. It has to be examined whether the resistance of high-strength steel joints can be based

638 on a plastic stress distribution.

639

640 Yu [19] also reported that the resistances of six full-width RHS-to-RHS T joints under brace in-plane bending  
641 were close to those of comparable RHS-to-RHS X joints (i.e., specimens of x10ie05, x10ie, x10ie2, x11ie2, x12i  
642 and x12ie2 in Table 16) with a maximum positive deviation of 5%. Thus, it is recommended to adopt the same  
643 resistance equations for full-width RHS-to-RHS X and T joints under brace in-plane bending.

644

645 Nagui [42] numerically examined the effects of chord sidewall convexity and thickness tolerance on full-width  
646 RHS-to-RHS T joints under brace in-plane bending. All the T joints had  $\eta^* = \eta = 1.0$ . Table 18 shows a comparison  
647 of the numerical resistances ( $M_{1u,ip}$ ) with the predicted resistances ( $M_{C,ip}$ ) obtained from Eq. (8) using  $\chi_C = 1.0$ . The  
648  $M_{1u,ip}/M_{C,ip}$  ratio becomes smaller for higher steel grades indicating more significant effects of the fabrication  
649 imperfections and more pronounced material effects. This further justifies the use of the material factor ( $C_f$ ) which  
650 could cover these effects for high-strength steel RHS joints.

651

## 652 **6. Discussion on RHS X and T joints under brace out-of-plane bending**

653

654 Table 19 tabulates the collated numerical results totalling eight full-width RHS-to-RHS X joints under brace out-  
655 of-plane bending reported by Yu [19]. The numerical resistances ( $M_{1u,op}$ ) were compared with the yield resistances  
656 ( $M_{y,op}$ ) derived from:

$$M_{y,op} = f_{y0} t_0 (b_0 - t_0) (h_1 + 5t_0) Q_f \quad (40)$$

657 It is shown that six joints reach the yield resistance ( $M_{y,op}$ ), and the resistance ratios ( $M_{1u,op}/M_{y,op}$ ) of three joints  
658 exceed 1.1.

659

660 The numerical results were adopted to evaluate the six design methods described in Section 3.1. The codified  
661 resistance equation (see Eq. (9)) was used. However, the  $\chi_C$  in Eq. (9) was replaced with  $\chi_{0.5}$  for the modified  
662 bearing-buckling method ( $M_{C,M,op}$ ),  $\chi_{Kuhn}$  (see Eq. (11)) for the Kuhn method ( $M_{Kuhn,op}$ ),  $\chi_{0.5}$  for the Yu method  
663 ( $M_{Yu,op}$ ) and  $\chi_{Lan}$  (see Eq. (28)) for the Lan method ( $M_{Lan,op}$ ). The term of  $\chi_C f_{y0}$  in Eq. (9) was replaced with  $f_{k,P}$  (see  
664 Eq. (33)) for the proposed modified bearing-buckling method ( $M_{P,M,op}$ ) and the Lan-Kuhn method ( $M_{P,LK,op}$ ). The  
665 Eurocode buckling curve c was conservatively used for all the RHS joints using hot-finished hollow sections.  
666 Table 20 shows that the proposed modified bearing-buckling method and the Lan-Kuhn method produce the lowest  
667 CoVs of 0.054 and 0.046, respectively. However, similar to the discussion in Section 5, it is currently difficult to  
668 draw generalised conclusions with respect to the design methods for the loading case of brace out-of-plane bending.  
669 This is because the database is small with three X joints having  $M_{1u,op}/M_{y,op} > 1.1$ , and most of the X joints examined  
670 reached the yield resistance. Fracture failure may occur due to the lower material ductility of high-strength steel.  
671 More tests, in particular for chord sidewall failure in high-strength steel joints, are thus required.

672

673 Yu [19] also numerically studied full-width RHS-to-RHS T joints under brace out-of-plane bending. These joints  
674 generally failed by distortion of the chord cross-section, and the corresponding joint resistance and stiffness largely  
675 depend on the unstiffened chord length. If chord distortion is prevented, the same resistance equation can be  
676 adopted for chord sidewall failure in RHS-to-RHS X and T joints under brace out-of-plane bending.

677

## 678 **7. Proposed design rules for RHS joints under brace axial compression**

679

680 More investigations on chord sidewall failure in high-strength steel RHS joints under brace axial tension, brace  
681 in-plane bending and brace out-of-plane bending are needed to assess the design methods comprehensively.  
682 Therefore, only design rules for chord sidewall failure in RHS X, T and Y joints under brace axial compression  
683 are proposed herein. The numerical study conducted by Yu [19] shows that the resistances of RHS-to-RHS T joints  
684 with  $n=0$  are slightly higher than those of comparable RHS-to-RHS X joints. The approach of only including a  
685 term of  $(1/\sin\theta_1)^{0.5}$  in the final resistance equation is suggested for RHS X, T and Y joints in this study because of  
686 the similar structural behaviour of these joints. Thus, the results of statistical analyses for X joints can be  
687 considered to be also representative for T and Y joints. The proposed design methods using the recommended  $C_f$   
688 factor can provide conservative resistance predictions for full-width RHS X joints with only one RHS brace welded  
689 to one chord side. For such RHS joints, the  $\chi_{P,M}$  or  $\chi_{P,LK}$  are appropriate for a welded plate (or similar) on the other  
690 chord side (class a joints), and the proposed  $\chi_{P,M1}$  or  $\chi_{P,LK1}$  are suitable for an unwelded support on the other chord  
691 side (class b joints). Hence, only the results of statistical analyses for RHS-to-RHS X joints under brace axial  
692 compression were adopted to evaluate the mean resistance to the design resistance for the proposed design methods.

693

694 AISC 360-16 [43] stipulates a reliability index of 3.0 for ductile welded hollow section joints and often adopts the  
695 simplified Eq. (41) from Ravindra and Galambos [44] to derive the resistance factor ( $\phi$ ). The beneficial overall  
696 effects of variations of geometric parameters and material properties are neglected in the calibration.

$$\phi = (\text{Mean}) e^{(-0.55)(3.0)(\text{CoV})} \quad (41)$$

697 Table 8 shows that the mean values of the  $N_{1w}/N_{P,M}$  and  $N_{1w}/N_{P,LK}$  ratios are 1.17 and 1.17, respectively, with  
698 corresponding CoVs of 0.091 and 0.095 for the evaluation against the test database. The corresponding obtained  
699  $\phi$  factors are 1.01 and 1.00. For the assessment against the numerical database (see Table 12), the mean values of  
700 the  $N_{1w}/N_{P,M}$  and  $N_{1w}/N_{P,LK}$  ratios are 1.23 and 1.24, respectively, with corresponding CoVs of 0.065 and 0.064.  
701 The corresponding derived  $\phi$  factors are 1.10 and 1.12. Thus, the smaller  $\phi$  factors obtained from the evaluation  
702 against the test results are governing and a rounded-off  $\phi$  factor of 1.0 can be adopted for the two proposed design  
703 methods. This indicates that the proposed modified bearing-buckling method and the Lan-Kuhn method produce  
704 equivalent nominal and design resistances.

705

706 Both the proposed modified bearing-buckling method and the Lan-Kuhn method adopt chord sidewall slenderness  
707 according to the conditions of chord sidewall end-resistant, an angle function of  $(1/\sin\theta_1)^{0.5}$  in the final resistance  
708 equation, an  $(\eta^*)^{-0.15}$  correction in the  $f_k$  function and a buckling curve c or equivalent. Tables 21-22 summarise  
709 the design rules using the two proposed design methods. It should be noted that for plate-to-RHS X joints and  
710 RHS-to-RHS X joints with  $h_1/h_0 \leq 0.25$ ,  $f_k = f_{y0}$  is suggested in line with Kuhn et al. [21]; however, the nominal  $f_{y0}$   
711 values should not exceed 460 MPa due to the lack of test data for high-strength steel RHS joints.

712

## 713 **8. Conclusions**

714

715 This study deals with the design of chord sidewall failure in rectangular hollow section (RHS) X, T and Y joints.

716 Test and numerical results reported in the literature for chord sidewall failure in RHS joints were collated. A wide  
717 range of geometric parameters, steel grades up to S960 and loading cases of brace axial loading, brace in-plane  
718 bending and brace out-of-plane bending were investigated. The effects of brace-to-chord height ratio ( $\eta^*$ ), brace  
719 angle ( $\theta_1$ ), chord stress ratio ( $n$ ) and steel grade were evaluated. The representative existing design approaches and  
720 two proposed design methods were evaluated against the compiled test and numerical results. Further required  
721 research on, in particular, high-strength steel RHS joints under brace axial tension, brace in-plane bending and  
722 brace out-of-plane bending, was discussed. The conclusions for the loading case of brace axial compression are  
723 summarised as follows:

724

725 (1) The effect of the  $\eta^*$  ratio on the joint resistance is pronounced and incorporating a correction term of  $(\eta^*)^{-0.15}$   
726 in the buckling stress function ( $f_k$ ) significantly reduces the scatter of resistance predictions.

727

728 (2) The approach of only including a term of  $(1/\sin\theta_1)^{0.5}$  in the final resistance equation can more accurately  
729 quantify the effect of brace angles and is recommended.

730

731 (3) The current codified chord stress function ( $Q_f$ ) becomes more conservative for large absolute values of the  $n$   
732 ratio and can be used to provide lower bound predictions for the chord stress effect.

733

734 (4) An equation for the material factor ( $C_f$ ) is suggested to consider the material effect; the rounded-off  $C_f$  values  
735 are 1.0, 0.95, 0.90, 0.85 and 0.80 for steel grades of S355, S460, S700, S900 and S960, respectively.

736

737 (5) The proposed modified bearing-buckling method and the simpler Lan-Kuhn method provide more consistent  
738 resistance predictions when compared with the existing design methods.

739

740 (6) Tables 21-22, which are based on the two proposed alternative design methods, summarise the proposed  
741 design rules for chord sidewall failure, which consider varying conditions of the chord sidewall end-restraint,  
742 with a resulting resistance factor ( $\phi$ ) of 1.0.

743

744

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746

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## References

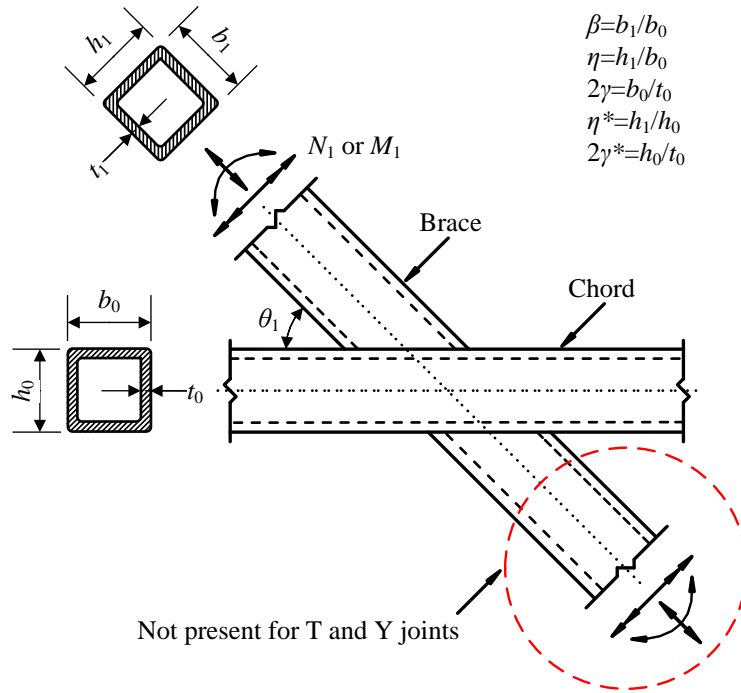
- [1] Czechowski J., Brodka J., Etude de la resistance statique des assemblages soudes en croix de profils creux rectangulaires. *Constr. Metallique* 3 (1977) 17-25.
- [2] Barentse J., Investigation into the static strength of welded T-joints made of rectangular hollow sections. Stevin Reports 6-76-23 and 6-77-7, Delft University of Technology, The Netherlands, 1977.
- [3] Brodka J., Szlendak J., Strength of cross joints in rectangular hollow sections. XXVI scientific conference of the civil and hydraulic engineering section of the Polish academy of science and of the science division of PZITB, 1980.
- [4] Kato B., Nishiyama I., The static strength of RR-joints with large b/B ratio. *IIW Doc. XV-459-80*, 1980.
- [5] Wardenier J., Davies G., The strength of predominantly statically loaded joints with a square or rectangular hollow section chord. *IIW Doc. XV-462-81*, 1981.
- [6] Eurocode 3 (EC3), Design of steel structures-Part 1-1: General rules and rules for buildings, EN 1993-1-1. European Committee for Standardization, CEN, Brussels, Belgium, 2005.
- [7] Eurocode 3 (EC3), Design of steel structures-Part 1-8: Design of joints, EN 1993-1-8. European Committee for Standardization, CEN, Brussels, Belgium, 2005.
- [8] Static design procedure for welded hollow-section joints-Recommendations, ISO/FDIS 14346. International Standardization Organization, ISO, Geneva, Switzerland, 2013.
- [9] Packer J.A., Wardenier J., Kurobane Y., Dutta D., Yeomans N., Design guide for rectangular hollow section (RHS) joints under predominantly static loading, 1st Ed., CIDECT, Verlag TUV Rheinland, Cologne, Germany, 1992.
- [10] Packer J.A., Wardenier J., Zhao X.L., van der Vegte G.J., Kurobane Y., Design guide for rectangular hollow section (RHS) joints under predominantly static loading, 2nd Ed., CIDECT, Geneva, Switzerland, 2009.
- [11] Design recommendations for hollow section joints-Predominantly statically loaded, 1st Ed., International Institute of Welding (IIW), Subcommittee XV-E, Rev. *IIW Doc. XV-491-81*, 1981.
- [12] Design recommendations for hollow section joints-Predominantly statically loaded, 2nd Ed., International Institute of Welding (IIW), Subcommittee XV-E, *IIW Doc. XV-701-89*, 1989.
- [13] Static design procedure for welded hollow section joints-Recommendations, 3rd Ed., International Institute of Welding (IIW), Subcommittee XV-E, *IIW Doc. XV-1402-12*, 2012.
- [14] Davies G., Packer J.A., Coutie M.G., The behavior of full width RHS cross joints. Proceedings of IIW international conference on welding of tubular structures, Boston, U.S.A, 1984.
- [15] Packer J.A., Web crippling of rectangular hollow sections. *J. Struct. Eng.* 110 (10) (1984) 2357-2373.
- [16] Giddings T.W., Wardenier J., The strength and behaviour of statically loaded welded connections in structural hollow sections, CIDECT Monograph 6, 1986.
- [17] Davies G., Roodbaraky K., The effect of angle on the strength of RHS joints. Proceedings of the international meeting on safety criteria in design of tubular structures, Tokyo, Japan, 1987.
- [18] Platt J.C., Sidewall behaviour in full-width rectangular hollow section joints. PhD thesis, University of Nottingham, Nottingham, U.K, 1984.
- [19] Yu Y., The static strength of uniplanar and multiplanar connections in rectangular hollow sections. PhD thesis,



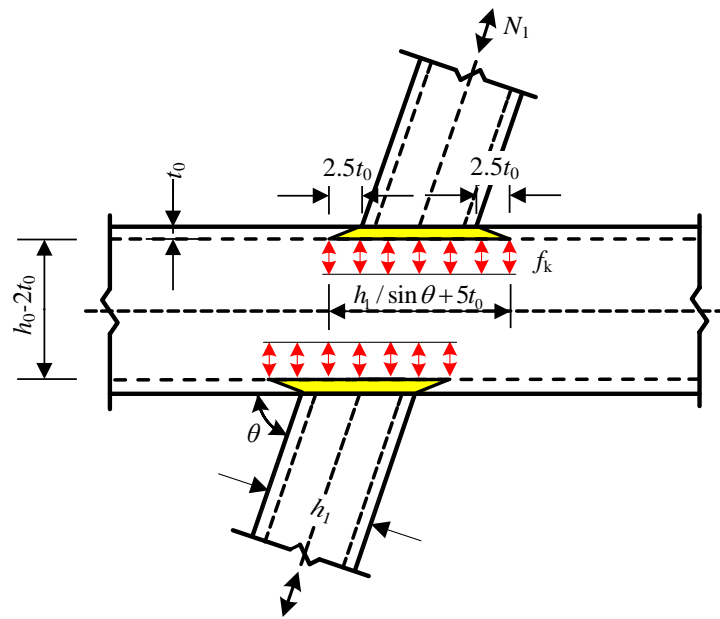
Delft University of Technology, Delft, The Netherlands, 1997.

- [20] Becque J., Cheng S., Sidewall buckling of equal-width RHS truss X-joints. *J. Struct. Eng.* 143(2) (2016) 04016179.
- [21] Kuhn J., Packer J.A., Fan Y.J., RHS webs under transverse compression. *Can. J. Civ. Eng.* 46 (2019) 810-827.
- [22] Wardenier J., Minutes of IIW Subcommission XV-E meeting, Singapore, IIW Doc. XV-E 482<sup>c</sup>-19, 2019.
- [23] Lan X.Y., Chan T.M., Young B., Structural behaviour and design of high strength steel RHS X-joints. *Eng. Struct.* 200 (2019) 109494.
- [24] Lan X.Y., Chan T.M., Young B., Testing, finite element analysis and design of high strength steel RHS T-joints. *Eng. Struct.* 227 (2020) 111184.
- [25] Eurocode 3 (EC3), Design of steel structures-Part 1-8: Design of joints, prEN 1993-1-8. European Committee for Standardization, CEN, Brussels, Belgium, 2019.
- [26] Wardenier J., Vegte G.J. van der, Liu, D.K., Chord stress function for rectangular hollow section X and T joints. Proceedings 17th International Offshore and Polar Engineering Conference, Lisbon, Portugal, 2007.
- [27] Lan X.Y., Chan T.M., Young B., Structural behaviour and design of chord plastification in high strength steel CHS X-joints. *Constr. Build. Mater.* 191 (2018) 1252-1267.
- [28] Lan X.Y., Chan T.M., Young B., Structural behaviour and design of high strength steel CHS T-joints. *Thin Wall. Struct.* <https://doi.org/10.1016/j.tws.2020.107215>.
- [29] Wardenier J., Hollow Section Joints. Delft University Press, The Netherlands, 1982.
- [30] Wardenier J., Lan X.Y., Packer J.A., Evaluation of design methods for chord sidewall failure in RHS joints using steel grades up to S960-state of the art. IIW Doc. XV-E-489-20, 2020.
- [31] Fan Y., RHS-to-RHS axially loaded X-connections offset towards an open chord end. Master thesis, University of Toronto, Toronto, Canada, 2017.
- [32] Feldmann M., Schillo N., Schaffrath S., Viridi K., Björk T., Tuominen N., Veljkovic M., Pavlovic M., Manoleas P., Heinisuo M., Mela K., Ongelin P., Valkonen I., Minkkinen J., Erkkilä J., Pétursson E., Clarin M., Seyr A., Horváth L., Kövesdi B., Turán P., Somodi B., Rules on high strength steel, Luxembourg, Publications Office of the European Union, 2016.
- [33] Pandey M., Young B., Structural performance of cold-formed high strength steel tubular X-joints under brace axial compression. *Eng. Struct.* 208 (2020) 109768.
- [34] Kim S.H., Lee C.H., Shin D.J., Chord stress effect in high-strength steel tubular X-joints. Proceedings 17th International Symposium on Tubular Structures, Singapore, 2019.
- [35] Zhao X.L., Deformation limit and ultimate strength of welded T-joints in cold-formed RHS sections. *J. Constr. Steel Res.* 53 (2000) 149-165.
- [36] Pandey M., Young B., Compression capacities of cold-formed high strength steel tubular T-joints. *J. Constr. Steel Res.* 162 (2019) 105650.
- [37] Poloni T., The effect of bearing length on transversely compressed RHS. Bachelor thesis, University of Toronto, Toronto, Canada, 1985.
- [38] De Koning C.H.M., Wardenier J., The static strength of welded joints between structural hollow sections or between structural hollow sections and H-sections, Part 2 - Joints between rectangular hollow sections. Delft University of Technology, Stevin report 6-84-19, Delft, The Netherlands, 1985.

- [39] Becque J., Wilkinson T., The capacity of grade C450 cold-formed rectangular hollow section T and X connections: an experimental investigation. *J. Constr. Steel Res.* 133 (2017) 345-359.
- [40] Björk T., Saastamoinen H., Capacity of CFRHS X joints made of double grade S420 steel. Proceedings 14th International Symposium on Tubular Structures, London, U.K., 2012.
- [41] Tuominen N., Björk T., Capacity of RHS-joints made of high strength steels. CIDECT Report 5BZ, Lappeenranta University of Technology, Finland, 2016.
- [42] Nagui A., A comparative analysis of the resistance of tubular joints. Master thesis, University of Bologna, Italy, 2016.
- [43] Specification for structural steel buildings. American Institute of Steel Construction (AISC), ANSI/AISC 360-16, Chicago, U.S.A, 2016.
- [44] Ravindra M.K., Galambos T.V., Load and resistance factor design for steel. *J. Struct. Div.* 104 (ST9) (1978) 1337-1353.



**Fig. 1.** Configurations and notations of RHS-to-RHS X, T and Y joints.



**Fig. 2.** Codified bearing-buckling model for chord side wall failure.

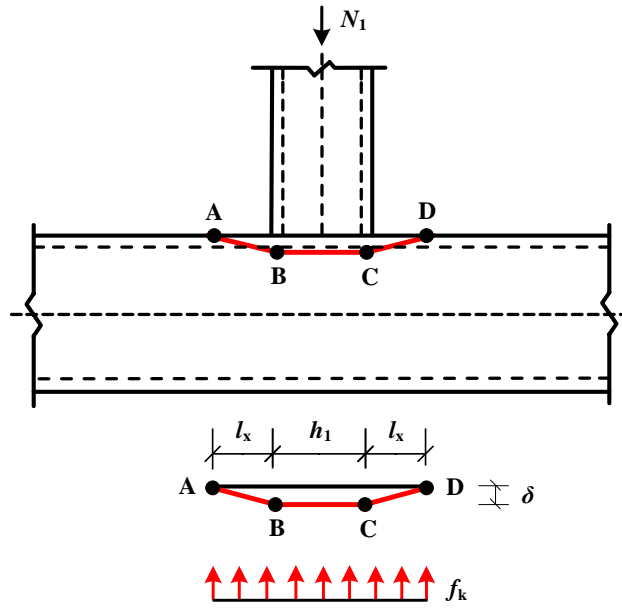
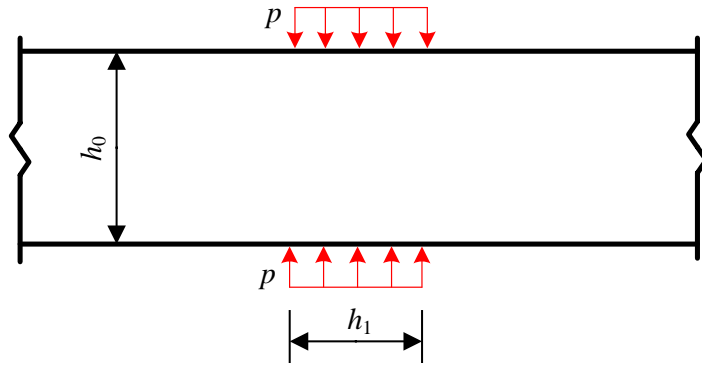
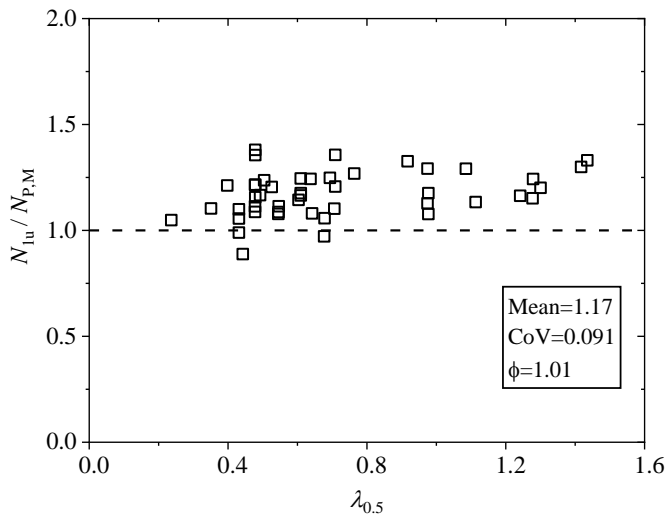


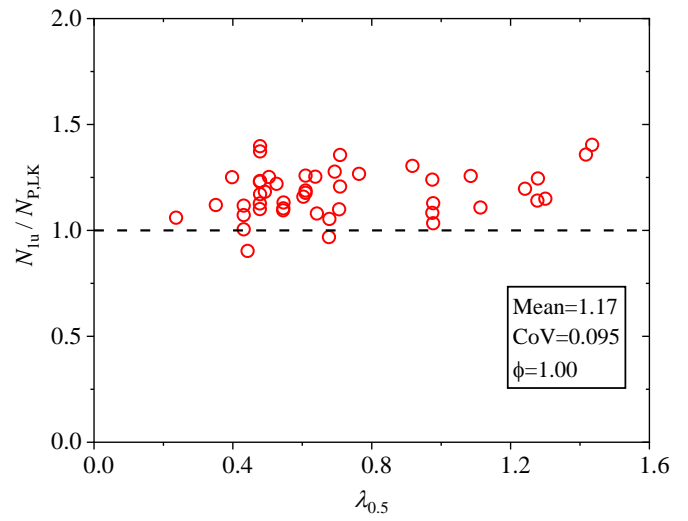
Fig. 3. Four-hinge yield line model proposed by Yu [19].



**Fig. 4.** Plate buckling model proposed by Lan et al. [23-24].

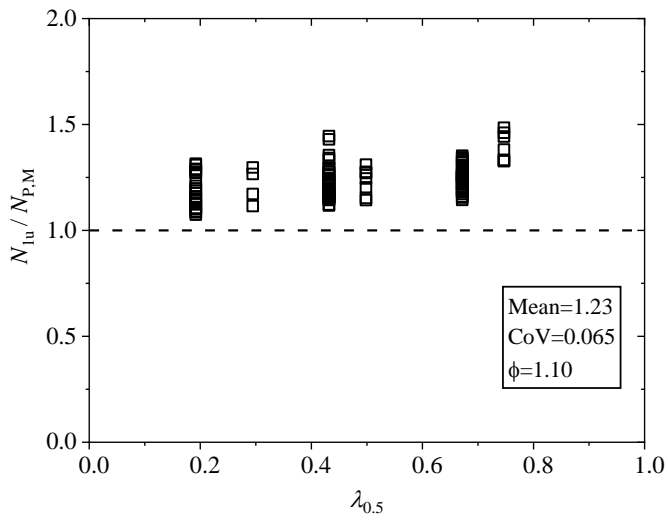


(a) Modified bearing-buckling method

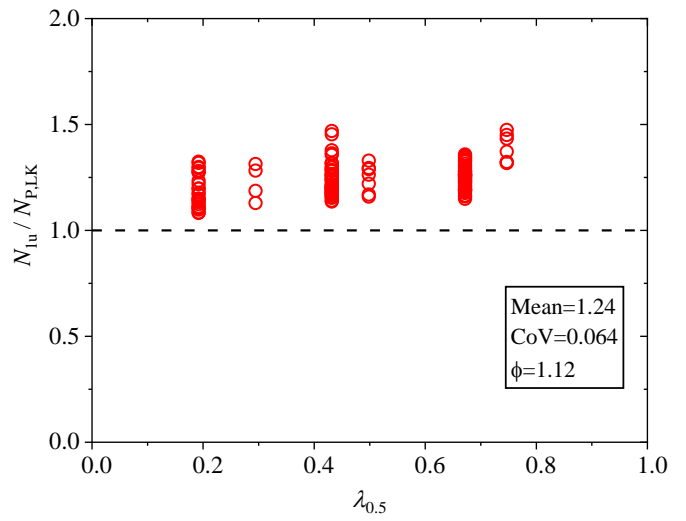


(b) Lan-Kuhn method

**Fig. 5.** Comparison of test resistances of 46 RHS-to-RHS joints under brace axial compression with those predicted by the proposed design methods, using the  $C_f$  factor.



(a) Modified bearing-buckling method



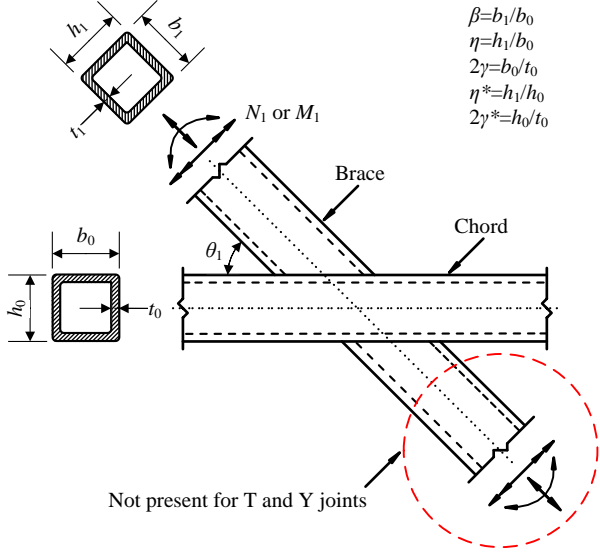
(b) Lan-Kuhn method

**Fig. 6.** Comparison of numerical resistances of 131 RHS-to-RHS joints under brace axial compression with those predicted by the proposed design methods, using the  $C_f$  factor.



**Table 1**

Codified design resistance for chord side wall failure in mild steel RHS joints with  $\beta=1.0$  [8, 10, 13].

T, Y and X joints	Brace axial loading		
 <p style="text-align: right;"> <math>\beta=b_1/b_0</math>  <math>\eta=h_1/b_0</math>  <math>2\gamma=b_0/t_0</math>  <math>\eta^*=h_1/h_0</math>  <math>2\gamma^*=h_0/t_0</math> </p> <p>Not present for T and Y joints</p>	$N_{1,Rd} = \frac{f_k t_0}{\sin \theta_1} \left( \frac{2h_1}{\sin \theta_1} + 10t_0 \right) Q_f$	<p><b>Tension:</b> <math>f_k = f_{y0}</math></p>	
		<p><b>Compression:</b>  <math>f_k = \chi_C f_{y0}</math> for T and Y joints  <math>f_k = 0.8 \chi_C f_{y0} \sin \theta_1</math> for X joints</p>	
	<p><b>Brace in-plane bending</b></p>		<p><math>f_k = f_{y0}</math> for T and Y joints  <math>f_k = 0.8 \chi_C f_{y0}</math> for X joints</p>
	$M_{ip,1,Rd} = 0.5 f_k t_0 (h_1 + 5t_0)^2 Q_f$		
	<p><b>Brace out-of-plane bending</b></p>		<p><math>f_k = \chi_C f_{y0}</math> for T and Y joints  <math>f_k = 0.8 \chi_C f_{y0}</math> for X joints</p>
	$M_{op,1,Rd} = f_k t_0 (b_0 - t_0) (h_1 + 5t_0) Q_f$		
<p><b>Parameters</b></p>			
<p>where <math>\chi_C</math> is the reduction factor for column buckling according to e.g., EN 1993-1-1 [6] using the relevant buckling curves and a normalised slenderness defined by:</p> $\lambda_c = \frac{3.46 \left( \frac{h_0}{t_0} - 2 \right) \sqrt{\frac{1}{\sin \theta_1}}}{\pi \sqrt{\frac{E}{f_{y0}}}}$			
<p><math>Q_f = (1 -  n )^{0.1}</math> with <math>n</math> in connecting chord face</p> <p><math>n = \frac{N_{0,Ed}}{N_{pl,0,Rd}} + \frac{M_{0,Ed}}{M_{pl,0,Rd}}</math> for class 1 or 2 chord cross-sections under chord compression stress and for chord cross-sections under chord tension stress</p>			
<p><b>Validity ranges</b></p>			
<p>steel grades up to S355; <math>b_0/t_0 \leq 40</math>, <math>h_0/t_0 \leq 40</math>, the chord cross-section shall be class 1 or 2 for the chord under compression stress; <math>0.5 \leq h_0/b_0 \leq 2.0</math>; <math>\theta_1 \geq 30^\circ</math>.</p>			

**Table 2**Comparison of buckling reduction factors for RHS joints with  $h_1/h_0=1.0$  and  $\theta_1=90^\circ$ .

$f_{y0}$ (MPa)	$h_0/t_0$	$\chi_{P,M}$	$\chi_{P,LK}$	$\chi_{Kuhn}$	$\chi_{Lan}$	$\chi_{P,LK}/\chi_{P,M}$	$\chi_{Kuhn}/\chi_{P,M}$	$\chi_{Lan}/\chi_{P,M}$
960	10	0.95	0.92	0.93	1.00	0.97	0.98	1.05
	20	0.74	0.73	0.72	0.86	0.98	0.97	1.16
	30	0.52	0.53	0.50	0.60	1.02	0.98	1.16
	35	0.42	0.43	0.40	0.47	1.02	0.94	1.12
	40	0.34	0.33	0.29	0.34	0.96	0.84	0.98
700	10	0.97	0.95	0.97	1.00	0.98	0.99	1.03
	20	0.80	0.78	0.78	0.94	0.98	0.98	1.17
	30	0.61	0.61	0.60	0.72	1.01	0.99	1.18
	40	0.43	0.45	0.41	0.49	1.04	0.96	1.14
460	10	1.00	0.98	1.00	1.00	0.99	1.00	1.00
	20	0.86	0.85	0.85	1.00	0.98	0.99	1.16
	30	0.71	0.71	0.70	0.84	1.00	0.99	1.19
	40	0.55	0.57	0.55	0.66	1.04	1.00	1.20
355	10	1.00	1.00	1.00	1.00	1.00	1.00	1.00
	20	0.89	0.88	0.89	1.00	0.99	0.99	1.12
	30	0.77	0.76	0.76	0.91	0.99	0.99	1.19
	40	0.62	0.64	0.63	0.75	1.03	1.00	1.20
235	10	1.00	1.02	1.00	1.00	1.02	1.00	1.00
	20	0.93	0.92	0.94	1.00	0.99	1.00	1.07
	30	0.83	0.83	0.83	1.00	0.99	1.00	1.20
	40	0.72	0.73	0.72	0.87	1.01	1.00	1.20
Max						1.04	1.00	1.20
Min						0.96	0.84	0.98
Mean						1.00	0.98	1.12
CoV						0.023	0.036	0.070

**Table 3**

Collated test results totalling 51 RHS-to-RHS X joints under brace axial compression.

Researcher/year	Specimen	$b_0$ (mm)	$h_0$ (mm)	$t_0$ (mm)	$b_1$ (mm)	$h_1$ (mm)	$t_1$ (mm)	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$\eta^*$	$2\gamma^*$	$N_{1u}$ (kN)	$N_{1u}/N_y$
Davies/1982	X(3)RR90	100.0	100.0	3.97	100.0	100.0	4.00	320	90	0	1.00	25.2	353	1.16
	X(4)RR45	100.0	100.0	3.93	100.0	100.0	4.00	320	45	0	1.00	25.4	372	1.04
Platt/1982	X-RR-90-A	100.0	100.2	4.20	100.0	100.0	4.00	432	90	0	1.00	23.9	391	0.89
	X-RR-60-A	99.8	100.1	4.20	100.0	100.0	4.00	432	60	0	1.00	23.8	410	0.87
	X-RR-45-A	100.1	100.0	4.20	100.0	100.0	4.00	432	45	0	1.00	23.8	450	0.86
	X-RR-90-B	98.7	100.0	4.00	100.0	50.0	5.00	311	90	0	0.50	25.0	209	1.20
	X-RR-60-B	98.9	100.0	4.10	100.0	50.0	5.00	311	60	0	0.50	24.4	218	1.13
	X-RR-45-B	99.2	100.0	4.00	100.0	50.0	5.00	311	45	0	0.50	25.0	244	1.18
	X-RR-90-C	250.0	251.1	6.50	250.0	250.0	6.30	237	90	0	1.00	38.6	680	0.78
	X-RR-60-C	250.4	250.7	6.50	250.0	250.0	6.30	237	60	0	1.00	38.6	672	0.72
X-RR-45-C	251.2	250.4	6.70	250.0	250.0	6.30	228	45	0	1.00	37.4	846	0.82	
Peksa/1982	10P	99.2	99.2	4.00	99.2	99.2	4.00	304	90	0	1.00	24.8	276	0.95
	11P	99.3	99.3	4.00	99.3	99.3	4.00	304	90	-0.43	1.00	24.8	271	0.93
	12P	99.3	99.3	4.00	99.3	99.3	4.00	304	90	-0.87	1.00	24.8	275	0.95
	13P	99.3	99.3	4.00	99.3	99.3	4.00	304	90	-0.87	1.00	24.8	280	0.97
	14P	99.2	99.2	4.00	99.2	99.2	4.00	304	90	-0.65	1.00	24.8	271	0.93
	15P	99.2	99.2	4.00	99.2	99.2	4.00	304	90	-0.22	1.00	24.8	263	0.91
Bettison/1982	5B	99.6	99.6	4.20	99.6	99.6	4.20	336	90	-0.71	1.00	23.7	313	0.92
	6B	99.8	99.8	4.10	99.8	99.8	4.10	336	90	-0.76	1.00	24.3	284	0.86
Poloni/1985	PWLR	102.4	252.7	4.44	102.4	252.7	4.44	388	90	0	1.00	56.9	438	0.46
Dixon/1983	DD1121	101.7	77.6	5.08	101.7	77.6	5.08	301	90	0	1.00	15.3	403	1.28
	DD1122	77.8	101.8	4.93	77.8	101.8	4.93	370	90	0	1.00	20.6	445	0.96
	DD1222	77.8	101.8	4.93	77.8	101.8	4.93	370	45	0	1.00	20.6	476	0.87
	DD1322	77.8	101.8	4.93	77.8	101.8	4.93	370	60	0	1.00	20.6	459	0.93
	DD2121	304.4	204.1	7.21	304.4	204.1	7.21	406	90	0	1.00	28.3	1315	0.93
	DD2122	204.1	304.4	7.21	204.1	304.4	7.21	406	90	0	1.00	42.2	1230	0.62
	DD2222	204.1	304.4	7.21	204.1	304.4	7.21	406	45	0	1.00	42.2	1675	0.71
	DD3121	203.2	153.6	4.83	203.2	153.6	4.83	392	90	0	1.00	31.8	649	0.97
	DD3122	153.6	203.2	4.83	153.6	203.2	4.83	412	90	0	1.00	42.1	530	0.59
	DD3221	203.2	153.6	4.83	203.2	153.6	4.83	392	44	0	1.00	31.8	693	0.86
	DD3222	153.6	203.2	4.83	153.6	203.2	4.83	412	44	0	1.00	42.1	694	0.64
	DD4123	254.1	254.1	9.35	254.1	254.1	9.35	406	90	0	1.00	27.2	2183	0.96
	DD4223	254.1	254.1	9.35	254.1	254.1	9.35	406	45	0	1.00	27.2	2429	0.90
DD4323	254.1	254.1	9.35	254.1	254.1	9.35	406	60	0	1.00	27.2	2215	0.90	
Cheng/2016	X1	100.5	100.3	2.92	100.2	100.3	2.73	330	90	0	1.00	34.3	176	0.79
	X2	100.4	100.1	3.84	100.4	100.2	3.69	330	90	0	1.00	26.1	302	1.00
	X3	100.3	99.8	4.89	100.1	99.9	4.70	400	90	0	1.00	20.4	373	0.77
	X4	99.6	99.6	5.80	99.8	99.7	5.46	370	90	0	1.00	17.2	560	1.01
	X5	99.9	99.7	7.92	100.1	99.6	7.68	345	90	0	1.00	12.6	783	1.03
	X6	149.8	250.0	5.00	150.1	150.1	4.76	463	90	0	0.60	50.0	409	0.50
	X7	150.2	150.2	5.86	150.5	150.4	5.86	451	90	0	1.00	25.6	828	0.87
	X9	300.0	400.0	7.92	300.3	300.3	7.97	481	90	0	0.75	50.5	1289	0.50
	Pandey/2020	X-100×50×4- 100×50×4	100.6	50.5	3.97	100.6	50.6	3.97	952	90	0	1.00	12.7	482
X-120×120×4- 120×120×4		121.6	121.7	3.93	121.4	121.8	3.92	971	90	0	1.00	31.0	567	0.53
X-140×140×4- 140×140×4		140.4	141.5	3.99	141.6	140.4	4.00	1008	90	0	0.99	35.5	484	0.37
X-120×120×3- 120×120×3		120.8	120.4	3.12	120.7	120.3	3.11	1038	90	0	1.00	38.6	317	0.36
X-80×80×4- 80×80×4		80.2	80.4	3.98	80.4	80.2	3.97	1004	90	0	1.00	20.2	595	0.74
X-120×120×4- 120×120×3		120.4	120.8	3.09	121.1	121.4	3.95	1038	90	0	1.00	39.1	318	0.36
Björk/2015	X5-S500	150.0	150.0	5.15	150.0	150.0	5.15	548	90	0	1.00	29.1	815	0.82
	X5-S700	150.0	150.0	5.06	150.0	150.0	5.06	762	90	0	1.00	29.6	935	0.69
	X5-S960	150.0	150.0	4.97	150.0	150.0	4.97	1080	90	0	1.00	30.2	808	0.43

**Table 4**Evaluation of design methods with the angle functions in  $\lambda_{0.5}, f_k$  and the final resistance equations.

Specimen	$f_{y0}$ (MPa)	$\theta_1$ (°)	$\eta^*$	$2\gamma^*$	$N_{1u}$ (kN)	$N_{1u}/N_{C,M}$	$N_{1u}/N_{Y_u}$	$N_{1u}/N_{Kuhn}$	$N_{1u}/N_{Lan}$	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
X(4)RR45	320	45	1.00	25.4	372	1.17	1.04	1.19	0.92	1.17	1.11
X-RR-90-A	432	90	1.00	23.9	391	1.09	1.02	1.11	0.92	1.09	1.11
X-RR-60-A	432	60	1.00	23.8	410	1.04	0.96	1.06	0.85	1.04	1.03
X-RR-45-A	432	45	1.00	23.8	450	1.01	0.90	1.03	0.79	1.01	0.95
X-RR-90-C	237	90	1.00	38.6	680	1.06	0.97	1.06	0.88	1.06	1.05
X-RR-60-C	237	60	1.00	38.6	672	0.96	0.87	0.96	0.77	0.96	0.92
X-RR-45-C	228	45	1.00	37.4	846	1.04	0.93	1.05	0.79	1.04	0.94
DD1122	370	90	1.00	20.6	445	1.10	1.07	1.10	0.96	1.10	1.11
DD1322	370	60	1.00	20.6	459	1.02	0.97	1.04	0.88	1.02	1.02
DD1222	370	45	1.00	20.6	476	0.93	0.85	0.94	0.77	0.93	0.89
DD4123	406	90	1.00	27.2	2183	1.23	1.14	1.24	1.03	1.23	1.24
DD4323	406	60	1.00	27.2	2215	1.14	1.04	1.16	0.93	1.14	1.11
DD4223	406	45	1.00	27.2	2429	1.11	0.99	1.13	0.85	1.11	1.02
DD3121	392	90	1.00	31.8	649	1.34	1.21	1.35	1.13	1.34	1.34
DD3221	392	44	1.00	31.8	693	1.18	1.02	1.19	0.87	1.18	1.04
DD2122	406	90	1.00	42.2	1230	1.11	1.06	1.10	0.92	1.11	1.07
DD2222	406	45	1.00	42.2	1675	1.35	1.24	1.37	0.92	1.35	1.06
DD3122	412	90	1.00	42.1	530	1.06	1.00	1.05	0.88	1.06	1.02
DD3222	412	44	1.00	42.1	694	1.23	1.12	1.26	0.83	1.23	0.96
Mean						1.11	1.02	1.13	0.89	1.11	1.05
CoV						0.103	0.103	0.105	0.101	0.103	0.104

Note: The material factor ( $C_f$ ) was not used for all design methods.

**Table 5**Evaluation of design methods only including a function of  $(1/\sin\theta_1)^{0.5}$  in the final resistance equation.

Specimen	$f_{y0}$ (MPa)	$\theta_1$ (°)	$\eta^*$	$2\gamma^*$	$N_{1u}$ (kN)	$N_{1u}/N_{C,M}$	$N_{1u}/N_{Y_u}$	$N_{1u}/N_{Kuhn}$	$N_{1u}/N_{Lan}$	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
X(4)RR45	320	45	1.00	25.4	372	1.24	1.16	1.25	1.04	1.24	1.25
X-RR-90-A	432	90	1.00	23.9	391	1.09	1.02	1.11	0.92	1.09	1.11
X-RR-60-A	432	60	1.00	23.8	410	1.06	1.00	1.08	0.90	1.06	1.08
X-RR-45-A	432	45	1.00	23.8	450	1.05	0.99	1.07	0.89	1.05	1.07
X-RR-90-C	237	90	1.00	38.6	680	1.06	0.97	1.06	0.88	1.06	1.05
X-RR-60-C	237	60	1.00	38.6	672	0.97	0.89	0.97	0.81	0.97	0.97
X-RR-45-C	228	45	1.00	37.4	846	1.08	0.99	1.08	0.90	1.08	1.08
DD1122	370	90	1.00	20.6	445	1.10	1.07	1.10	0.96	1.10	1.11
DD1322	370	60	1.00	20.6	459	1.05	1.03	1.06	0.93	1.05	1.07
DD1222	370	45	1.00	20.6	476	0.99	0.96	0.99	0.87	0.99	1.00
DD4123	406	90	1.00	27.2	2183	1.23	1.14	1.24	1.03	1.23	1.24
DD4323	406	60	1.00	27.2	2215	1.16	1.08	1.17	0.98	1.16	1.17
DD4223	406	45	1.00	27.2	2429	1.15	1.07	1.16	0.97	1.15	1.16
DD3121	392	90	1.00	31.8	649	1.34	1.21	1.35	1.13	1.34	1.34
DD3221	392	44	1.00	31.8	693	1.19	1.07	1.21	1.00	1.19	1.19
DD2122	406	90	1.00	42.2	1230	1.11	1.06	1.10	0.92	1.11	1.07
DD2222	406	45	1.00	42.2	1675	1.27	1.21	1.26	1.06	1.27	1.22
DD3122	412	90	1.00	42.1	530	1.06	1.00	1.05	0.88	1.06	1.02
DD3222	412	44	1.00	42.1	694	1.16	1.09	1.15	0.96	1.16	1.11
Mean						1.12	1.05	1.13	0.95	1.12	1.12
CoV						0.086	0.079	0.087	0.082	0.086	0.085

Note: The material factor ( $C_f$ ) was not used for all design methods.

**Table 6**

Evaluation of the chord stress effect against test results of eight RHS-to-RHS X joints.

Specimen	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$\eta^*$	$N_{1u}$ (kN)	$N_{1u}/N_{C,M}$	$N_{1u}/N_{Y_u}$	$N_{1u}/N_{Kuhn}$	$N_{1u}/N_{Lan}$	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
10P	304	90	0	1.00	276	1.11	1.04	1.12	0.95	1.11	1.13
11P	304	90	-0.43	1.00	271	1.16	1.08	1.16	0.99	1.16	1.17
12P	304	90	-0.87	1.00	275	1.36	1.27	1.36	1.16	1.36	1.37
13P	304	90	-0.87	1.00	280	1.38	1.29	1.39	1.18	1.38	1.40
14P	304	90	-0.65	1.00	271	1.21	1.14	1.22	1.04	1.21	1.23
15P	304	90	-0.22	1.00	263	1.09	1.02	1.09	0.93	1.09	1.10
5B	336	90	-0.71	1.00	313	1.22	1.14	1.23	1.04	1.22	1.23
6B	336	90	-0.76	1.00	284	1.17	1.09	1.18	0.99	1.17	1.18
Mean						1.21	1.13	1.22	1.03	1.21	1.23
CoV						0.088	0.088	0.088	0.089	0.088	0.088

Note: The codified chord stress function ( $Q_f$ ) was adopted for all design methods and the material factor ( $C_f$ ) equals 1.0 for all the joints.

**Table 7**Evaluation of design methods, without using the  $C_f$  factor, against test results of 46 screened RHS-to-RHS X joints under brace axial compression.

Specimen	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$\eta^*$	$N_{1u}$ (kN)	$N_{1u}/N_{C,M}$	$N_{1u}/N_{Yu}$	$N_{1u}/N_{Kuhn}$	$N_{1u}/N_{Lan}$	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
X(4)RR45	320	45	0	1.00	372	1.24	1.16	1.25	1.04	1.24	1.25
X-RR-90-A	432	90	0	1.00	391	1.09	1.02	1.11	0.92	1.09	1.11
X-RR-60-A	432	60	0	1.00	410	1.06	1.00	1.08	0.90	1.06	1.08
X-RR-45-A	432	45	0	1.00	450	1.05	0.99	1.07	0.89	1.05	1.07
X-RR-90-C	237	90	0	1.00	680	1.06	0.97	1.06	0.88	1.06	1.05
X-RR-60-C	237	60	0	1.00	672	0.97	0.89	0.97	0.81	0.97	0.97
X-RR-45-C	228	45	0	1.00	846	1.08	0.99	1.08	0.90	1.08	1.08
10P	304	90	0	1.00	276	1.11	1.04	1.12	0.95	1.11	1.13
11P	304	90	-0.43	1.00	271	1.16	1.08	1.16	0.99	1.16	1.17
12P	304	90	-0.87	1.00	275	1.36	1.27	1.36	1.16	1.36	1.37
13P	304	90	-0.87	1.00	280	1.38	1.29	1.39	1.18	1.38	1.40
14P	304	90	-0.65	1.00	271	1.21	1.14	1.22	1.04	1.21	1.23
15P	304	90	-0.22	1.00	263	1.09	1.02	1.09	0.93	1.09	1.10
5B	336	90	-0.71	1.00	313	1.22	1.14	1.23	1.04	1.22	1.23
6B	336	90	-0.76	1.00	284	1.17	1.09	1.18	0.99	1.17	1.18
PWLR	388	90	0	1.00	438	1.19	1.16	1.25	1.06	1.19	1.14
DD1122	370	90	0	1.00	445	1.09	1.07	1.10	0.96	1.09	1.11
DD1222	370	45	0	1.00	476	0.99	0.96	0.99	0.87	0.99	1.00
DD1322	370	60	0	1.00	459	1.05	1.03	1.06	0.93	1.05	1.07
DD2121	406	90	0	1.00	1315	1.23	1.09	1.24	1.03	1.23	1.24
DD2122	406	90	0	1.00	1230	1.11	1.06	1.10	0.92	1.11	1.07
DD2222	406	45	0	1.00	1675	1.27	1.21	1.26	1.06	1.27	1.22
DD3121	392	90	0	1.00	649	1.34	1.21	1.35	1.13	1.34	1.34
DD3122	412	90	0	1.00	530	1.06	1.00	1.05	0.88	1.06	1.02
DD3221	392	44	0	1.00	693	1.19	1.07	1.21	1.00	1.19	1.19
DD3222	412	44	0	1.00	694	1.16	1.09	1.15	0.96	1.16	1.11
DD4123	406	90	0	1.00	2183	1.23	1.14	1.24	1.03	1.23	1.24
DD4223	406	45	0	1.00	2429	1.15	1.07	1.16	0.97	1.15	1.16
DD4323	406	60	0	1.00	2215	1.16	1.08	1.17	0.98	1.16	1.17
X1	330	90	0	1.00	176	1.10	1.02	1.11	0.92	1.10	1.10
X2	330	90	0	1.00	302	1.20	1.12	1.22	1.01	1.20	1.22
X3	400	90	0	1.00	373	0.88	0.83	0.89	0.77	0.88	0.89
X4	370	90	0	1.00	560	1.10	1.06	1.10	1.01	1.10	1.11
X5	345	90	0	1.00	783	1.05	1.05	1.04	1.03	1.05	1.06
X6	463	90	0	0.60	409	1.22	1.13	1.25	0.83	1.13	1.07
X7	451	90	0	1.00	828	1.11	1.04	1.13	0.94	1.11	1.13
X9	481	90	0	0.75	1289	1.25	1.15	1.31	0.94	1.20	1.15
X-100×50×4- 100×50×4	952	90	0	1.00	482	1.01	0.90	1.03	0.91	1.01	1.04
X-120×120×4- 120×120×4	971	90	0	1.00	567	1.07	0.99	1.10	0.92	1.07	1.04
X-140×140×4- 140×140×4	1008	90	0	0.99	484	0.94	0.87	1.02	0.86	0.94	0.93
X-120×120×3- 120×120×3	1038	90	0	1.00	317	1.05	0.97	1.26	1.08	1.05	1.10
X-80×80×4- 80×80×4	1004	90	0	1.00	595	1.02	0.97	1.05	0.88	1.02	1.04
X-120×120×4- 120×120×3	1038	90	0	1.00	318	1.07	0.99	1.31	1.13	1.07	1.13
X5-S500	548	90	0	1.00	815	1.20	1.11	1.21	1.01	1.20	1.20
X5-S700	762	90	0	1.00	935	1.17	1.09	1.19	1.00	1.17	1.16
X5-S960	1080	90	0	1.00	808	0.90	0.84	0.93	0.79	0.90	0.88
Mean						1.13	1.05	1.15	0.97	1.12	1.13
CoV						0.098	0.096	0.098	0.097	0.097	0.097

Note: The approach of only including a term of  $(1/\sin\theta_1)^{0.5}$  in the final resistance equation was adopted for all design methods.

**Table 8**Evaluation of design methods, using the  $C_f$  factor, against test results of 46 screened RHS-to-RHS X joints under brace axial compression.

Specimen	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$\eta^*$	$N_{1u}$ (kN)	$N_{1u}/N_{C,M}$	$N_{1u}/N_{Y_u}$	$N_{1u}/N_{Kuhn}$	$N_{1u}/N_{Lan}$	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
X(4)RR45	320	45	0	1.00	372	1.24	1.16	1.25	1.04	1.24	1.25
X-RR-90-A	432	90	0	1.00	391	1.11	1.05	1.13	0.94	1.11	1.13
X-RR-60-A	432	60	0	1.00	410	1.09	1.02	1.10	0.92	1.09	1.10
X-RR-45-A	432	45	0	1.00	450	1.08	1.01	1.09	0.91	1.08	1.09
X-RR-90-C	237	90	0	1.00	680	1.06	0.97	1.06	0.88	1.06	1.05
X-RR-60-C	237	60	0	1.00	672	0.97	0.89	0.97	0.81	0.97	0.97
X-RR-45-C	228	45	0	1.00	846	1.08	0.99	1.08	0.90	1.08	1.08
10P	304	90	0	1.00	276	1.11	1.04	1.12	0.95	1.11	1.13
11P	304	90	-0.43	1.00	271	1.16	1.08	1.16	0.99	1.16	1.17
12P	304	90	-0.87	1.00	275	1.36	1.27	1.36	1.16	1.36	1.37
13P	304	90	-0.87	1.00	280	1.38	1.29	1.39	1.18	1.38	1.40
14P	304	90	-0.65	1.00	271	1.21	1.14	1.22	1.04	1.21	1.23
15P	304	90	-0.22	1.00	263	1.09	1.02	1.09	0.93	1.09	1.10
5B	336	90	-0.71	1.00	313	1.22	1.14	1.23	1.04	1.22	1.23
6B	336	90	-0.76	1.00	284	1.17	1.09	1.18	0.99	1.17	1.18
PWLR	388	90	0	1.00	438	1.20	1.17	1.26	1.07	1.20	1.15
DD1122	370	90	0	1.00	445	1.10	1.07	1.11	0.97	1.10	1.12
DD1222	370	45	0	1.00	476	0.99	0.97	1.00	0.87	0.99	1.00
DD1322	370	60	0	1.00	459	1.06	1.03	1.06	0.93	1.06	1.07
DD2121	406	90	0	1.00	1315	1.24	1.10	1.26	1.05	1.24	1.25
DD2122	406	90	0	1.00	1230	1.13	1.07	1.12	0.94	1.13	1.08
DD2222	406	45	0	1.00	1675	1.29	1.23	1.28	1.07	1.29	1.24
DD3121	392	90	0	1.00	649	1.36	1.22	1.37	1.14	1.36	1.36
DD3122	412	90	0	1.00	530	1.08	1.01	1.07	0.90	1.08	1.03
DD3221	392	44	0	1.00	693	1.21	1.08	1.22	1.01	1.21	1.21
DD3222	412	44	0	1.00	694	1.18	1.11	1.17	0.98	1.18	1.13
DD4123	406	90	0	1.00	2183	1.24	1.16	1.26	1.05	1.24	1.26
DD4223	406	45	0	1.00	2429	1.16	1.08	1.18	0.98	1.16	1.18
DD4323	406	60	0	1.00	2215	1.18	1.09	1.19	0.99	1.18	1.19
X1	330	90	0	1.00	176	1.10	1.02	1.11	0.92	1.10	1.10
X2	330	90	0	1.00	302	1.20	1.12	1.22	1.01	1.20	1.22
X3	400	90	0	1.00	373	0.89	0.84	0.90	0.78	0.89	0.90
X4	370	90	0	1.00	560	1.10	1.06	1.11	1.02	1.10	1.12
X5	345	90	0	1.00	783	1.05	1.05	1.04	1.03	1.05	1.06
X6	463	90	0	0.60	409	1.26	1.16	1.29	0.86	1.16	1.11
X7	451	90	0	1.00	828	1.14	1.07	1.16	0.97	1.14	1.16
X9	481	90	0	0.75	1289	1.30	1.19	1.36	0.98	1.24	1.19
X-100×50×4- 100×50×4	952	90	0	1.00	482	1.21	1.08	1.24	1.09	1.21	1.25
X-120×120×4- 120×120×4	971	90	0	1.00	567	1.29	1.20	1.32	1.11	1.29	1.26
X-140×140×4- 140×140×4	1008	90	0	0.99	484	1.15	1.06	1.25	1.05	1.15	1.14
X-120×120×3- 120×120×3	1038	90	0	1.00	317	1.30	1.20	1.56	1.33	1.30	1.36
X-80×80×4- 80×80×4	1004	90	0	1.00	595	1.25	1.18	1.29	1.07	1.25	1.28
X-120×120×4- 120×120×3	1038	90	0	1.00	318	1.33	1.22	1.62	1.40	1.33	1.41
X5-S500	548	90	0	1.00	815	1.27	1.18	1.28	1.07	1.27	1.27
X5-S700	762	90	0	1.00	935	1.33	1.23	1.34	1.12	1.33	1.30
X5-S960	1080	90	0	1.00	808	1.13	1.05	1.17	0.99	1.13	1.11
Mean						1.17	1.10	1.20	1.01	1.17	1.17
CoV						0.092	0.086	0.116	0.116	0.091	0.095

Note: The approach of only including a term of  $(1/\sin\theta_1)^{0.5}$  in the final resistance equation was adopted for all design methods.



**Table 9**

Collated numerical results totalling 173 RHS-to-RHS X joints under brace axial compression.

Researcher/year	Specimen	$b_0$ (mm)	$h_0$ (mm)	$t_0$ (mm)	$b_1$ (mm)	$h_1$ (mm)	$t_1$ (mm)	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$\eta^*$	$2\gamma^*$	$N_{1u}$ (kN)	$N_{1u}/N_y$
Yu/1997	x10ae05	150	150	10.00	150	75	10.00	355	90	0	0.50	15	1019	1.15
	x10ae	150	150	10.00	150	150	10.00	355	90	0	1.00	15	1507	1.06
	x10ae2	150	150	10.00	150	300	10.00	355	90	0	2.00	15	2498	1.01
	x11ae05	150	150	6.25	150	75	6.25	355	90	0	0.50	24	509	1.08
	x11a	150	150	6.25	150	150	6.25	355	90	0	1.00	24	776	0.96
	x11ae2	150	150	6.25	150	300	6.25	355	90	0	2.00	24	1427	0.97
	x12ae05	150	150	4.29	150	75	4.29	355	90	0	0.50	35	302	1.03
	x12ae	150	150	4.29	150	150	4.29	355	90	0	1.00	35	482	0.92
	x12ae2	150	150	4.29	150	300	4.29	355	90	0	2.00	35	816	0.83
	x10a-0	150	150	10.00	150	150	10.00	355	90	0	1.00	15	1611	1.13
	x10a-0.4	150	150	10.00	150	150	10.00	355	90	-0.40	1.00	15	1587	1.12
	x10a-0.6	150	150	10.00	150	150	10.00	355	90	-0.60	1.00	15	1562	1.10
	x10a-0.8	150	150	10.00	150	150	10.00	355	90	-0.80	1.00	15	1492	1.05
	x11a-0	150	150	6.25	150	150	6.25	355	90	0	1.00	24	817	1.02
	x11a-0.4	150	150	6.25	150	150	6.25	355	90	-0.40	1.00	24	803	1.00
	x11a-0.6	150	150	6.25	150	150	6.25	355	90	-0.60	1.00	24	787	0.98
	x11a-0.8	150	150	6.25	150	150	6.25	355	90	-0.80	1.00	24	757	0.94
	x12a-0	150	150	4.29	150	150	4.29	355	90	0	1.00	35	502	0.96
	x12a-0.4	150	150	4.29	150	150	4.29	355	90	-0.40	1.00	35	498	0.95
	x12a-0.6	150	150	4.29	150	150	4.29	355	90	-0.60	1.00	35	484	0.93
	x12a-0.8	150	150	4.29	150	150	4.29	355	90	-0.80	1.00	35	459	0.88
Kuhn/2019	Para_1	200	80	8.00	200	400	11.00	398	90	0	5.00	10	3239	1.16
	Para_26	200	80	8.00	200	400	11.00	398	90	-0.25	5.00	10	3252	1.16
	Para_51	200	80	8.00	200	400	11.00	398	90	-0.50	5.00	10	3252	1.16
	Para_76	200	80	8.00	200	400	11.00	398	90	-0.75	5.00	10	3237	1.15
	Para_101	200	80	8.00	200	400	11.00	398	90	0.25	5.00	10	3225	1.15
	Para_126	200	80	8.00	200	400	11.00	398	90	0.50	5.00	10	3186	1.14
	Para_151	200	80	8.00	200	400	11.00	398	90	0.75	5.00	10	3135	1.12
	Para_2	200	160	8.00	200	400	11.00	398	90	0	2.50	20	2641	0.94
	Para_27	200	160	8.00	200	400	11.00	398	90	-0.25	2.50	20	2676	0.95
	Para_52	200	160	8.00	200	400	11.00	398	90	-0.50	2.50	20	2687	0.96
	Para_77	200	160	8.00	200	400	11.00	398	90	-0.75	2.50	20	2673	0.95
	Para_102	200	160	8.00	200	400	11.00	398	90	0.25	2.50	20	2608	0.93
	Para_127	200	160	8.00	200	400	11.00	398	90	0.50	2.50	20	2530	0.90
	Para_152	200	160	8.00	200	400	11.00	398	90	0.75	2.50	20	2392	0.85
	Para_6	200	80	8.00	200	200	11.00	398	90	0	2.50	10	1742	1.14
	Para_31	200	80	8.00	200	200	11.00	398	90	-0.25	2.50	10	1747	1.14
	Para_56	200	80	8.00	200	200	11.00	398	90	-0.50	2.50	10	1747	1.14
	Para_106	200	80	8.00	200	200	11.00	398	90	0.25	2.50	10	1737	1.14
	Para_131	200	80	8.00	200	200	11.00	398	90	0.50	2.50	10	1720	1.13
	Para_156	200	80	8.00	200	200	11.00	398	90	0.75	2.50	10	1694	1.11
	P_1	80	80	8.00	80	160	11.00	398	90	0	2.00	10	1423	1.12
	P_26	80	80	8.00	80	160	11.00	398	90	-0.25	2.00	10	1434	1.13
	P_51	80	80	8.00	80	160	11.00	398	90	-0.50	2.00	10	1433	1.12
	P_76	80	80	8.00	80	160	11.00	398	90	-0.75	2.00	10	1422	1.12
	P_101	80	80	8.00	80	160	11.00	398	90	0.25	2.00	10	1401	1.10
	P_126	80	80	8.00	80	160	11.00	398	90	0.50	2.00	10	1360	1.07
	P_151	80	80	8.00	80	160	11.00	398	90	0.75	2.00	10	1292	1.01
	P_2	160	160	8.00	160	320	11.00	398	90	0	2.00	20	2201	0.96
	P_27	160	160	8.00	160	320	11.00	398	90	-0.25	2.00	20	2228	0.97
	P_52	160	160	8.00	160	320	11.00	398	90	-0.50	2.00	20	2237	0.98
	P_77	160	160	8.00	160	320	11.00	398	90	-0.75	2.00	20	2237	0.98
	P_102	160	160	8.00	160	320	11.00	398	90	0.25	2.00	20	2170	0.95
	P_127	160	160	8.00	160	320	11.00	398	90	0.50	2.00	20	2088	0.91
P_152	160	160	8.00	160	320	11.00	398	90	0.75	2.00	20	1969	0.86	
P_3	240	240	8.00	240	480	11.00	398	90	0	2.00	30	2504	0.76	
P_28	240	240	8.00	240	480	11.00	398	90	-0.25	2.00	30	2549	0.77	
P_53	240	240	8.00	240	480	11.00	398	90	-0.50	2.00	30	2561	0.77	

P_78	240	240	8.00	240	480	11.00	398	90	-0.75	2.00	30	2532	0.76
P_103	240	240	8.00	240	480	11.00	398	90	0.25	2.00	30	2465	0.74
P_128	240	240	8.00	240	480	11.00	398	90	0.50	2.00	30	2408	0.73
P_153	240	240	8.00	240	480	11.00	398	90	0.75	2.00	30	2257	0.68
Para_3	200	240	8.00	200	400	11.00	398	90	0	1.67	30	2198	0.78
Para_28	200	240	8.00	200	400	11.00	398	90	-0.25	1.67	30	2235	0.80
Para_53	200	240	8.00	200	400	11.00	398	90	-0.50	1.67	30	2236	0.80
Para_78	200	240	8.00	200	400	11.00	398	90	-0.75	1.67	30	2217	0.79
Para_103	200	240	8.00	200	400	11.00	398	90	0.25	1.67	30	2166	0.77
Para_128	200	240	8.00	200	400	11.00	398	90	0.50	1.67	30	2112	0.75
Para_153	200	240	8.00	200	400	11.00	398	90	0.75	1.67	30	1967	0.70
Para_7	200	160	8.00	200	200	11.00	398	90	0	1.25	20	1511	0.99
Para_32	200	160	8.00	200	200	11.00	398	90	-0.25	1.25	20	1526	1.00
Para_57	200	160	8.00	200	200	11.00	398	90	-0.50	1.25	20	1516	0.99
Para_82	200	160	8.00	200	200	11.00	398	90	-0.75	1.25	20	1502	0.98
Para_107	200	160	8.00	200	200	11.00	398	90	0.25	1.25	20	1498	0.98
Para_132	200	160	8.00	200	200	11.00	398	90	0.50	1.25	20	1456	0.95
Para_157	200	160	8.00	200	200	11.00	398	90	0.75	1.25	20	1384	0.91
Para_11	200	80	8.00	200	100	11.00	398	90	0	1.25	10	987	1.11
Para_36	200	80	8.00	200	100	11.00	398	90	-0.25	1.25	10	984	1.10
Para_61	200	80	8.00	200	100	11.00	398	90	-0.50	1.25	10	974	1.09
Para_86	200	80	8.00	200	100	11.00	398	90	-0.75	1.25	10	957	1.07
Para_111	200	80	8.00	200	100	11.00	398	90	0.25	1.25	10	990	1.11
Para_136	200	80	8.00	200	100	11.00	398	90	0.50	1.25	10	987	1.11
Para_161	200	80	8.00	200	100	11.00	398	90	0.75	1.25	10	975	1.09
P_6	80	80	8.00	80	80	11.00	398	90	0	1.00	10	823	1.08
P_31	80	80	8.00	80	80	11.00	398	90	-0.25	1.00	10	826	1.08
P_56	80	80	8.00	80	80	11.00	398	90	-0.50	1.00	10	817	1.07
P_81	80	80	8.00	80	80	11.00	398	90	-0.75	1.00	10	795	1.04
P_106	80	80	8.00	80	80	11.00	398	90	0.25	1.00	10	810	1.06
P_131	80	80	8.00	80	80	11.00	398	90	0.50	1.00	10	775	1.01
P_156	80	80	8.00	80	80	11.00	398	90	0.75	1.00	10	706	0.92
P_7	160	160	8.00	160	160	11.00	398	90	0	1.00	20	1279	1.00
P_32	160	160	8.00	160	160	11.00	398	90	-0.25	1.00	20	1287	1.01
P_57	160	160	8.00	160	160	11.00	398	90	-0.50	1.00	20	1274	1.00
P_82	160	160	8.00	160	160	11.00	398	90	-0.75	1.00	20	1250	0.98
P_107	160	160	8.00	160	160	11.00	398	90	0.25	1.00	20	1266	0.99
P_132	160	160	8.00	160	160	11.00	398	90	0.50	1.00	20	1223	0.96
P_157	160	160	8.00	160	160	11.00	398	90	0.75	1.00	20	1145	0.90
P_8	240	240	8.00	240	240	11.00	398	90	0	1.00	30	1547	0.87
P_33	240	240	8.00	240	240	11.00	398	90	-0.25	1.00	30	1547	0.87
P_58	240	240	8.00	240	240	11.00	398	90	-0.50	1.00	30	1530	0.86
P_83	240	240	8.00	240	240	11.00	398	90	-0.75	1.00	30	1490	0.84
P_108	240	240	8.00	240	240	11.00	398	90	0.25	1.00	30	1551	0.87
P_133	240	240	8.00	240	240	11.00	398	90	0.50	1.00	30	1507	0.84
P_158	240	240	8.00	240	240	11.00	398	90	0.75	1.00	30	1418	0.80
Para_8	200	240	8.00	200	200	11.00	398	90	0	0.83	30	1385	0.91
Para_33	200	240	8.00	200	200	11.00	398	90	-0.25	0.83	30	1380	0.90
Para_58	200	240	8.00	200	200	11.00	398	90	-0.50	0.83	30	1352	0.88
Para_83	200	240	8.00	200	200	11.00	398	90	-0.75	0.83	30	1302	0.85
Para_108	200	240	8.00	200	200	11.00	398	90	0.25	0.83	30	1385	0.91
Para_133	200	240	8.00	200	200	11.00	398	90	0.50	0.83	30	1340	0.88
Para_158	200	240	8.00	200	200	11.00	398	90	0.75	0.83	30	1246	0.82
Para_12	200	160	8.00	200	100	11.00	398	90	0	0.63	20	933	1.05
Para_37	200	160	8.00	200	100	11.00	398	90	-0.25	0.63	20	936	1.05
Para_62	200	160	8.00	200	100	11.00	398	90	-0.50	0.63	20	918	1.03
Para_87	200	160	8.00	200	100	11.00	398	90	-0.75	0.63	20	881	0.99
Para_112	200	160	8.00	200	100	11.00	398	90	0.25	0.63	20	930	1.04
Para_137	200	160	8.00	200	100	11.00	398	90	0.50	0.63	20	908	1.02
Para_162	200	160	8.00	200	100	11.00	398	90	0.75	0.63	20	863	0.97
Para_16	200	80	8.00	200	50	11.00	398	90	0	0.63	10	626	1.09
Para_41	200	80	8.00	200	50	11.00	398	90	-0.25	0.63	10	618	1.08
Para_66	200	80	8.00	200	50	11.00	398	90	-0.50	0.63	10	603	1.05

Para_91	200	80	8.00	200	50	11.00	398	90	-0.75	0.63	10	581	1.01
Para_116	200	80	8.00	200	50	11.00	398	90	0.25	0.63	10	632	1.10
Para_141	200	80	8.00	200	50	11.00	398	90	0.50	0.63	10	633	1.10
Para_166	200	80	8.00	200	50	11.00	398	90	0.75	0.63	10	629	1.10
P_11	80	80	8.00	80	40	11.00	398	90	0	0.50	10	596	1.17
P_36	80	80	8.00	80	40	11.00	398	90	-0.25	0.50	10	596	1.17
P_61	80	80	8.00	80	40	11.00	398	90	-0.50	0.50	10	585	1.15
P_86	80	80	8.00	80	40	11.00	398	90	-0.75	0.50	10	558	1.10
P_111	80	80	8.00	80	40	11.00	398	90	0.25	0.50	10	576	1.13
P_136	80	80	8.00	80	40	11.00	398	90	0.50	0.50	10	540	1.06
P_161	80	80	8.00	80	40	11.00	398	90	0.75	0.50	10	476	0.93
P_12	160	160	8.00	160	80	11.00	398	90	0	0.50	20	824	1.08
P_37	160	160	8.00	160	80	11.00	398	90	-0.25	0.50	20	822	1.08
P_62	160	160	8.00	160	80	11.00	398	90	-0.50	0.50	20	800	1.05
P_87	160	160	8.00	160	80	11.00	398	90	-0.75	0.50	20	758	0.99
P_112	160	160	8.00	160	80	11.00	398	90	0.25	0.50	20	820	1.07
P_137	160	160	8.00	160	80	11.00	398	90	0.50	0.50	20	799	1.05
P_162	160	160	8.00	160	80	11.00	398	90	0.75	0.50	20	749	0.98
P_13	240	240	8.00	240	120	11.00	398	90	0	0.50	30	1039	1.02
P_38	240	240	8.00	240	120	11.00	398	90	-0.25	0.50	30	1025	1.01
P_63	240	240	8.00	240	120	11.00	398	90	-0.50	0.50	30	984	0.97
P_88	240	240	8.00	240	120	11.00	398	90	-0.75	0.50	30	914	0.90
P_138	240	240	8.00	240	120	11.00	398	90	0.50	0.50	30	1020	1.00
P_163	240	240	8.00	240	120	11.00	398	90	0.75	0.50	30	964	0.95
Para_13	200	240	8.00	200	100	11.00	398	90	0	0.42	30	949	1.06
Para_38	200	240	8.00	200	100	11.00	398	90	-0.25	0.42	30	936	1.05
Para_63	200	240	8.00	200	100	11.00	398	90	-0.50	0.42	30	892	1.00
Para_88	200	240	8.00	200	100	11.00	398	90	-0.75	0.42	30	816	0.91
Para_113	200	240	8.00	200	100	11.00	398	90	0.25	0.42	30	958	1.07
Para_138	200	240	8.00	200	100	11.00	398	90	0.50	0.42	30	935	1.05
Para_163	200	240	8.00	200	100	11.00	398	90	0.75	0.42	30	878	0.98
Para_17	200	160	8.00	200	50	11.00	398	90	0	0.31	20	651	1.14
Para_42	200	160	8.00	200	50	11.00	398	90	-0.25	0.31	20	637	1.11
Para_67	200	160	8.00	200	50	11.00	398	90	-0.50	0.31	20	610	1.06
Para_92	200	160	8.00	200	50	11.00	398	90	-0.75	0.31	20	569	0.99
Para_117	200	160	8.00	200	50	11.00	398	90	0.25	0.31	20	661	1.15
Para_142	200	160	8.00	200	50	11.00	398	90	0.50	0.31	20	661	1.15
Para_18	200	240	8.00	200	50	11.00	398	90	0	0.21	30	662	1.15
Para_43	200	240	8.00	200	50	11.00	398	90	-0.25	0.21	30	646	1.13
Para_68	200	240	8.00	200	50	11.00	398	90	-0.50	0.21	30	609	1.06
Para_93	200	240	8.00	200	50	11.00	398	90	-0.75	0.21	30	551	0.96
Para_118	200	240	8.00	200	50	11.00	398	90	0.25	0.21	30	687	1.20
Para_143	200	240	8.00	200	50	11.00	398	90	0.50	0.21	30	693	1.21
Para_168	200	240	8.00	200	50	11.00	398	90	0.75	0.21	30	677	1.18
Para_167	200	160	8.00	200	50	11.00	398	90	0.75	0.31	20	643	1.12
P_18	240	240	8.00	240	60	11.00	398	90	0	0.25	30	730	1.15
P_43	240	240	8.00	240	60	11.00	398	90	-0.25	0.25	30	712	1.12
P_68	240	240	8.00	240	60	11.00	398	90	-0.50	0.25	30	676	1.06
P_93	240	240	8.00	240	60	11.00	398	90	-0.75	0.25	30	613	0.96
P_118	240	240	8.00	240	60	11.00	398	90	0.25	0.25	30	746	1.17
P_143	240	240	8.00	240	60	11.00	398	90	0.50	0.25	30	753	1.18
P_168	240	240	8.00	240	60	11.00	398	90	0.75	0.25	30	743	1.17

**Table 10**Numerical results of 22 RHS-to-RHS X joints selected to examine the effect of  $\eta^*$  ratio.

Specimen	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$\eta^*$	$N_{1u}$ (kN)	$N_{1u}/N_{C,M}$	$N_{1u}/N_{Yu}$	$N_{1u}/N_{Kuhn}$	$N_{1u}/N_{Lan}$	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
x10ae	355	90	0	1.00	1019	1.11	1.09	1.11	1.06	1.11	1.13
x10ae2	355	90	0	2.00	1507	1.06	1.04	1.05	1.01	1.17	1.19
x11ae05	355	90	0	0.50	2498	1.28	1.15	1.29	1.08	1.15	1.17
x11a	355	90	0	1.00	509	1.14	1.07	1.15	0.96	1.14	1.16
x11ae2	355	90	0	2.00	776	1.15	1.11	1.16	1.06	1.28	1.29
x12ae05	355	90	0	0.50	1427	1.48	1.29	1.49	1.10	1.33	1.32
x12ae	355	90	0	1.00	302	1.33	1.22	1.33	1.11	1.33	1.32
x12ae2	355	90	0	2.00	482	1.20	1.15	1.20	1.19	1.33	1.32
Para_2	398	90	0	2.50	1507	1.08	1.04	1.09	1.01	1.24	1.26
P_2	398	90	0	2.00	2498	1.10	1.07	1.11	1.00	1.22	1.24
P_3	398	90	0	2.00	509	1.03	0.99	1.04	1.00	1.14	1.15
Para_3	398	90	0	1.67	776	1.07	1.03	1.08	1.00	1.16	1.16
Para_7	398	90	0	1.25	1427	1.14	1.06	1.15	1.00	1.18	1.20
P_6	398	90	0	1.00	302	1.09	1.13	1.09	1.09	1.09	1.10
P_7	398	90	0	1.00	482	1.15	1.10	1.16	1.02	1.15	1.17
P_8	398	90	0	1.00	816	1.18	1.10	1.20	1.00	1.18	1.19
Para_8	398	90	0	0.83	2641	1.24	1.16	1.25	1.01	1.20	1.21
Para_12	398	90	0	0.63	2201	1.20	1.08	1.21	1.06	1.12	1.14
Para_16	398	90	0	0.63	2504	1.11	0.93	1.11	1.11	1.11	1.11
P_12	398	90	0	0.50	2198	1.24	1.14	1.25	1.09	1.12	1.14
P_13	398	90	0	0.50	1511	1.39	1.22	1.41	1.06	1.25	1.26
Para_13	398	90	0	0.42	823	1.45	1.30	1.47	1.08	1.27	1.28
Mean							1.19	1.11	1.20	1.05	1.20
CoV							0.105	0.080	0.106	0.051	0.064

Note: The material factor ( $C_f$ ) was used for all the design methods.

**Table 11**

Numerical results of 10 RHS-to-RHS X joints selected to examine the chord stress effect.

Specimen	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$\eta^*$	$N_{1u}$ (kN)	$N_{1u}/N_{C,M}$	$N_{1u}/N_{Yu}$	$N_{1u}/N_{Kuhn}$	$N_{1u}/N_{Lan}$	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
x10a-0.6	355	90	-0.6	1.00	1562	1.27	1.24	1.26	1.21	1.27	1.28
x10a-0.8	355	90	-0.8	1.00	1492	1.30	1.27	1.29	1.23	1.30	1.31
x11a-0	355	90	0	1.00	817	1.20	1.13	1.21	1.02	1.20	1.22
x11a-0.4	355	90	-0.4	1.00	803	1.24	1.17	1.26	1.05	1.24	1.26
x11a-0.6	355	90	-0.6	1.00	787	1.27	1.19	1.28	1.07	1.27	1.29
x11a-0.8	355	90	-0.8	1.00	757	1.31	1.23	1.32	1.11	1.31	1.33
x12a-0	355	90	0	1.00	502	1.38	1.27	1.39	1.16	1.38	1.37
x12a-0.4	355	90	-0.4	1.00	498	1.44	1.33	1.45	1.21	1.44	1.43
x12a-0.6	355	90	-0.6	1.00	484	1.46	1.35	1.47	1.22	1.46	1.45
x12a-0.8	355	90	-0.8	1.00	459	1.48	1.37	1.49	1.24	1.48	1.47
Mean						1.34	1.25	1.34	1.15	1.34	1.34
CoV						0.074	0.063	0.073	0.073	0.074	0.064

Note: The material factor ( $C_f$ ) equals 1.0 for all the joints.

**Table 12**Evaluation of design methods, using the  $C_f$  factor, against numerical results of 131 screened RHS-to-RHS X joints under brace axial compression.

Specimen	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$\eta^*$	$N_{1u}$ (kN)	$N_{1u}/N_{C,M}$	$N_{1u}/N_{Yu}$	$N_{1u}/N_{Kuhn}$	$N_{1u}/N_{Lan}$	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
x10ae	355	90	0	1.00	1507	1.11	1.09	1.11	1.06	1.11	1.13
x10ae2	355	90	0	2.00	2498	1.06	1.04	1.05	1.01	1.17	1.19
x11ae05	355	90	0	0.50	509	1.28	1.15	1.29	1.08	1.15	1.17
x11a	355	90	0	1.00	776	1.14	1.07	1.15	0.96	1.14	1.16
x11ae2	355	90	0	2.00	1427	1.15	1.11	1.16	1.06	1.28	1.29
x12ae05	355	90	0	0.50	302	1.48	1.29	1.49	1.10	1.33	1.32
x12ae	355	90	0	1.00	482	1.33	1.22	1.33	1.11	1.33	1.32
x12ae2	355	90	0	2.00	816	1.20	1.15	1.20	1.19	1.33	1.32
x10a-0.6	355	90	-0.60	1.00	1562	1.27	1.24	1.26	1.21	1.27	1.28
x10a-0.8	355	90	-0.80	1.00	1492	1.30	1.27	1.29	1.23	1.30	1.31
x11a-0	355	90	0	1.00	817	1.20	1.13	1.21	1.02	1.20	1.22
x11a-0.4	355	90	-0.40	1.00	803	1.24	1.17	1.26	1.05	1.24	1.26
x11a-0.6	355	90	-0.60	1.00	787	1.27	1.19	1.28	1.07	1.27	1.29
x11a-0.8	355	90	-0.80	1.00	757	1.31	1.23	1.32	1.11	1.31	1.33
x12a-0	355	90	0	1.00	502	1.38	1.27	1.39	1.16	1.38	1.37
x12a-0.4	355	90	-0.40	1.00	498	1.44	1.33	1.45	1.21	1.44	1.43
x12a-0.6	355	90	-0.60	1.00	484	1.46	1.35	1.47	1.22	1.46	1.45
x12a-0.8	355	90	-0.80	1.00	459	1.48	1.37	1.49	1.24	1.48	1.47
Para_2	398	90	0	2.50	2641	1.08	1.04	1.09	1.01	1.24	1.26
Para_27	398	90	-0.25	2.50	2676	1.13	1.09	1.14	1.05	1.30	1.32
Para_52	398	90	-0.50	2.50	2687	1.18	1.14	1.19	1.10	1.36	1.38
Para_77	398	90	-0.75	2.50	2673	1.26	1.21	1.27	1.17	1.45	1.47
Para_102	398	90	0.25	2.50	2608	1.10	1.06	1.11	1.03	1.26	1.28
Para_127	398	90	0.50	2.50	2530	1.11	1.07	1.12	1.04	1.28	1.30
Para_152	398	90	0.75	2.50	2392	1.13	1.09	1.14	1.05	1.29	1.31
P_101	398	90	0.25	2.00	1401	1.15	1.17	1.15	1.15	1.27	1.28
P_126	398	90	0.50	2.00	1360	1.16	1.18	1.16	1.16	1.29	1.29
P_151	398	90	0.75	2.00	1292	1.18	1.21	1.18	1.18	1.31	1.32
P_2	398	90	0	2.00	2201	1.10	1.07	1.11	1.00	1.22	1.24
P_27	398	90	-0.25	2.00	2228	1.15	1.12	1.16	1.04	1.28	1.30
P_52	398	90	-0.50	2.00	2237	1.20	1.17	1.21	1.09	1.33	1.36
P_77	398	90	-0.75	2.00	2237	1.29	1.25	1.30	1.17	1.43	1.45
P_102	398	90	0.25	2.00	2170	1.12	1.09	1.13	1.01	1.24	1.26
P_127	398	90	0.50	2.00	2088	1.12	1.09	1.13	1.02	1.25	1.27
P_152	398	90	0.75	2.00	1969	1.13	1.10	1.14	1.03	1.26	1.28
P_3	398	90	0	2.00	2504	1.03	0.99	1.04	1.00	1.14	1.15
P_28	398	90	-0.25	2.00	2549	1.08	1.04	1.09	1.05	1.20	1.20
P_53	398	90	-0.50	2.00	2561	1.13	1.08	1.14	1.10	1.25	1.26
P_78	398	90	-0.75	2.00	2532	1.20	1.15	1.21	1.16	1.33	1.33
P_103	398	90	0.25	2.00	2465	1.04	1.00	1.06	1.01	1.16	1.16
P_128	398	90	0.50	2.00	2408	1.06	1.02	1.07	1.03	1.18	1.18
P_153	398	90	0.75	2.00	2257	1.07	1.02	1.08	1.04	1.18	1.19
Para_3	398	90	0	1.67	2198	1.07	1.03	1.08	1.00	1.16	1.16
Para_28	398	90	-0.25	1.67	2235	1.12	1.08	1.13	1.04	1.21	1.21
Para_53	398	90	-0.50	1.67	2236	1.17	1.12	1.18	1.09	1.26	1.26
Para_78	398	90	-0.75	1.67	2217	1.24	1.19	1.25	1.15	1.34	1.34
Para_103	398	90	0.25	1.67	2166	1.09	1.05	1.10	1.01	1.17	1.18
Para_128	398	90	0.50	1.67	2112	1.10	1.06	1.11	1.02	1.19	1.19
Para_153	398	90	0.75	1.67	1967	1.10	1.06	1.11	1.02	1.19	1.19
Para_7	398	90	0	1.25	1511	1.14	1.06	1.15	1.00	1.18	1.20
Para_32	398	90	-0.25	1.25	1526	1.18	1.10	1.19	1.04	1.22	1.24
Para_57	398	90	-0.50	1.25	1516	1.22	1.14	1.23	1.08	1.26	1.29
Para_82	398	90	-0.75	1.25	1502	1.30	1.21	1.31	1.14	1.34	1.36
Para_107	398	90	0.25	1.25	1498	1.16	1.09	1.17	1.02	1.20	1.22
Para_132	398	90	0.50	1.25	1456	1.17	1.10	1.18	1.03	1.21	1.23
Para_157	398	90	0.75	1.25	1384	1.20	1.12	1.21	1.05	1.24	1.26
Para_36	398	90	-0.25	1.25	984	1.15	1.03	1.15	1.15	1.19	1.20
Para_61	398	90	-0.50	1.25	974	1.19	1.06	1.19	1.19	1.23	1.23

Para_86	398	90	-0.75	1.25	957	1.25	1.12	1.25	1.25	1.29	1.30
Para_161	398	90	0.75	1.25	975	1.27	1.14	1.27	1.27	1.31	1.32
P_6	398	90	0	1.00	823	1.09	1.13	1.09	1.09	1.09	1.10
P_31	398	90	-0.25	1.00	826	1.13	1.17	1.13	1.13	1.13	1.13
P_56	398	90	-0.50	1.00	817	1.16	1.20	1.16	1.16	1.16	1.17
P_81	398	90	-0.75	1.00	795	1.21	1.25	1.21	1.21	1.21	1.22
P_106	398	90	0.25	1.00	810	1.10	1.14	1.10	1.10	1.10	1.11
P_131	398	90	0.50	1.00	775	1.10	1.14	1.10	1.10	1.10	1.11
P_156	398	90	0.75	1.00	706	1.07	1.11	1.07	1.07	1.07	1.08
P_7	398	90	0	1.00	1279	1.15	1.10	1.16	1.02	1.15	1.17
P_32	398	90	-0.25	1.00	1287	1.20	1.14	1.21	1.05	1.20	1.22
P_57	398	90	-0.50	1.00	1274	1.23	1.17	1.24	1.09	1.23	1.25
P_82	398	90	-0.75	1.00	1250	1.30	1.23	1.31	1.14	1.30	1.32
P_107	398	90	0.25	1.00	1266	1.18	1.12	1.19	1.03	1.18	1.20
P_132	398	90	0.50	1.00	1223	1.18	1.12	1.19	1.04	1.18	1.20
P_157	398	90	0.75	1.00	1145	1.19	1.13	1.20	1.04	1.19	1.21
P_8	398	90	0	1.00	1547	1.18	1.10	1.20	1.00	1.18	1.19
P_33	398	90	-0.25	1.00	1547	1.22	1.13	1.23	1.03	1.22	1.22
P_58	398	90	-0.50	1.00	1530	1.25	1.16	1.27	1.06	1.25	1.26
P_83	398	90	-0.75	1.00	1490	1.31	1.21	1.32	1.10	1.31	1.31
P_108	398	90	0.25	1.00	1551	1.22	1.13	1.23	1.03	1.22	1.23
P_133	398	90	0.50	1.00	1507	1.24	1.15	1.25	1.04	1.24	1.24
P_158	398	90	0.75	1.00	1418	1.25	1.16	1.26	1.05	1.25	1.25
Para_8	398	90	0	0.83	1385	1.24	1.16	1.25	1.01	1.20	1.21
Para_33	398	90	-0.25	0.83	1380	1.27	1.19	1.28	1.03	1.23	1.24
Para_58	398	90	-0.50	0.83	1352	1.29	1.21	1.31	1.06	1.26	1.26
Para_83	398	90	-0.75	0.83	1302	1.33	1.25	1.35	1.09	1.30	1.30
Para_108	398	90	0.25	0.83	1385	1.27	1.19	1.29	1.04	1.24	1.24
Para_133	398	90	0.50	0.83	1340	1.28	1.20	1.30	1.05	1.25	1.25
Para_158	398	90	0.75	0.83	1246	1.28	1.19	1.29	1.04	1.24	1.25
Para_12	398	90	0	0.63	933	1.20	1.08	1.21	1.06	1.12	1.14
Para_37	398	90	-0.25	0.63	936	1.24	1.11	1.25	1.09	1.16	1.18
Para_62	398	90	-0.50	0.63	918	1.27	1.13	1.28	1.12	1.18	1.20
Para_87	398	90	-0.75	0.63	881	1.30	1.17	1.32	1.15	1.22	1.24
Para_112	398	90	0.25	0.63	930	1.23	1.10	1.25	1.09	1.15	1.17
Para_137	398	90	0.50	0.63	908	1.25	1.12	1.27	1.10	1.17	1.19
Para_162	398	90	0.75	0.63	863	1.28	1.14	1.29	1.12	1.19	1.21
Para_16	398	90	0	0.63	626	1.11	0.93	1.11	1.11	1.11	1.11
Para_41	398	90	-0.25	0.63	618	1.12	0.95	1.12	1.12	1.12	1.12
Para_66	398	90	-0.50	0.63	603	1.14	0.96	1.14	1.14	1.14	1.14
Para_91	398	90	-0.75	0.63	581	1.18	0.99	1.18	1.18	1.18	1.18
Para_116	398	90	0.25	0.63	632	1.15	0.97	1.15	1.15	1.15	1.15
Para_141	398	90	0.50	0.63	633	1.20	1.01	1.20	1.20	1.20	1.20
Para_166	398	90	0.75	0.63	629	1.27	1.08	1.27	1.27	1.27	1.27
P_86	398	90	-0.75	0.50	558	1.27	1.35	1.27	1.27	1.27	1.27
P_136	398	90	0.50	0.50	540	1.15	1.21	1.15	1.15	1.15	1.15
P_161	398	90	0.75	0.50	476	1.09	1.15	1.09	1.09	1.09	1.09
P_12	398	90	0	0.50	824	1.24	1.14	1.25	1.09	1.12	1.14
P_37	398	90	-0.25	0.50	822	1.27	1.17	1.28	1.12	1.15	1.17
P_62	398	90	-0.50	0.50	800	1.29	1.19	1.30	1.14	1.16	1.18
P_87	398	90	-0.75	0.50	758	1.31	1.20	1.32	1.15	1.18	1.20
P_112	398	90	0.25	0.50	820	1.27	1.17	1.28	1.12	1.14	1.16
P_137	398	90	0.50	0.50	799	1.29	1.18	1.30	1.13	1.16	1.18
P_162	398	90	0.75	0.50	749	1.29	1.19	1.30	1.14	1.17	1.19
P_13	398	90	0	0.50	1039	1.39	1.22	1.41	1.06	1.25	1.26
P_38	398	90	-0.25	0.50	1025	1.41	1.24	1.43	1.07	1.27	1.28
P_63	398	90	-0.50	0.50	984	1.41	1.24	1.43	1.07	1.27	1.28
P_88	398	90	-0.75	0.50	914	1.41	1.24	1.42	1.07	1.27	1.27
P_138	398	90	0.50	0.50	1020	1.46	1.29	1.48	1.11	1.32	1.33
P_163	398	90	0.75	0.50	964	1.48	1.30	1.50	1.13	1.34	1.34
Para_13	398	90	0	0.42	949	1.45	1.30	1.47	1.08	1.27	1.28
Para_38	398	90	-0.25	0.42	936	1.47	1.32	1.49	1.09	1.29	1.30
Para_63	398	90	-0.50	0.42	892	1.46	1.31	1.48	1.09	1.28	1.29

Para_88	398	90	-0.75	0.42	816	1.43	1.28	1.45	1.06	1.26	1.26
Para_113	398	90	0.25	0.42	958	1.51	1.35	1.52	1.12	1.32	1.33
Para_138	398	90	0.50	0.42	935	1.53	1.37	1.55	1.14	1.34	1.35
Para_163	398	90	0.75	0.42	878	1.54	1.38	1.56	1.15	1.35	1.36
Para_67	398	90	-0.50	0.31	610	1.31	1.11	1.32	1.15	1.15	1.15
Para_92	398	90	-0.75	0.31	569	1.31	1.11	1.32	1.15	1.15	1.15
Para_68	398	90	-0.50	0.21	609	1.55	1.31	1.57	1.15	1.23	1.23
Para_93	398	90	-0.75	0.21	551	1.51	1.27	1.52	1.12	1.19	1.20
P_68	398	90	-0.50	0.25	676	1.55	1.27	1.57	1.15	1.26	1.27
P_93	398	90	-0.75	0.25	613	1.51	1.24	1.52	1.12	1.23	1.23
Mean						1.24	1.15	1.25	1.10	1.23	1.24
CoV						0.102	0.082	0.104	0.061	0.065	0.064

Note: The material factor ( $C_f$ ) was used for all the design methods.



**Table 13**

Collated test results totalling 24 RHS X joints with only one RHS brace welded to the chord and under brace axial compression.

Researcher/year	Specimen	$b_0$ (mm)	$h_0$ (mm)	$t_0$ (mm)	$b_1$ (mm)	$h_1$ (mm)	$t_1$ (mm)	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$2\gamma^*$	$\eta^*$	$N_{1u}$ (kN)	$N_{1u}/N_y$
Barentse/1977	T-RR-A-A-1	101.4	101.4	6.23	101.4	70.0	6.23	299	90	0	16.3	0.69	417	1.11
	T-RR-A-A-7	101.3	101.3	4.03	100.2	70.0	3.77	326	90	0	25.1	0.69	210	0.89
	T-RR-A-A-10	100.4	100.4	2.88	100.4	70.0	2.88	299	90	0	34.9	0.70	112	0.77
	T-RR-E-A-91	101.3	50.9	6.27	101.3	50.9	6.27	322	90	0	8.1	1.00	453	1.36
	T-RR-E-A-94	101.8	50.9	4.73	101.6	50.8	4.90	338	90	0	10.8	1.00	298	1.25
	T-RR-E-A-97	101.3	51.0	3.35	101.3	51.0	3.35	293	90	0	15.2	1.00	150	1.13
	T-RR-E-A-112	102.1	151.3	6.23	101.4	70.0	6.18	289	90	0	24.3	0.46	394	1.08
	T-RR-E-A-119	101.2	151.4	4.82	101.4	70.0	6.23	293	90	0	31.4	0.46	256	0.96
	T-RR-E-A-126	80.6	119.7	3.05	80.2	70.0	3.00	398	90	0	39.2	0.58	142	0.69
	T-RR-C-A-46	100.2	100.2	3.77	100.0	25.0	3.08	349	90	0	26.6	0.25	180	1.16
	T-RR-C-A-50	100.8	100.8	3.36	100.0	25.0	3.02	290	90	0	30.0	0.25	152	1.37
	T-RR-C-A-41	101.4	101.4	6.18	100.9	70.0	3.92	298	90	0	16.4	0.69	394	1.06
	T-RR-C-A-45	101.0	101.0	4.00	100.9	70.0	3.92	343	90	0	25.3	0.69	248	1.00
	T-RR-C-A-49	100.8	100.8	3.36	99.8	70.0	4.00	290	90	0	30.0	0.69	163	0.96
Zhao/2000	S1B1C11	51.0	102.0	4.90	51.0	51.0	4.90	409	90	0	20.8	0.50	316	1.04
	S1B1C12	51.0	102.0	3.20	51.0	51.0	4.90	343	90	0	31.9	0.50	163	1.11
	S1B2C21	102.0	102.0	9.50	102.0	102.0	8.00	445	90	0	10.7	1.00	1207	0.95
	S1B2C22	102.0	102.0	6.30	102.0	102.0	8.00	432	90	0	16.2	1.00	652	0.90
Pandey/2019	TF-100x50x4- 100x50x4	100.6	50.6	3.96	100.6	50.6	3.97	952	90	0	12.8	1.00	494	0.93
	TF-120x120x4- 120x120x4	121.6	121.6	3.91	121.6	121.7	3.91	971	90	0	31.1	1.00	558	0.52
	TF-140x140x4- 140x140x4	141.6	140.3	3.97	141.7	140.4	4.00	1008	90	0	35.3	1.00	544	0.42
	TF-120x120x3- 120x120x3	120.9	120.3	3.12	120.8	120.3	3.11	1038	90	0	38.5	1.00	369	0.42
Fan/2017	X-1.0-32-7000	203.6	203.6	5.96	204.0	204.0	11.67	404	90	0	34.2	1.00	653	0.58
	X-1.0-21-5500	203.1	203.1	8.85	204.0	204.0	11.67	418	90	0	22.9	1.00	1264	0.69

**Table 14**Evaluation of proposed design methods, using the  $C_f$  factor, against test results of eight RHS X joints classified as class a.

Specimen	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$\eta^*$	$N_{1u}$ (kN)	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
T-RR-A-A-7	326	90	0	0.69	210	1.00	1.01
T-RR-A-A-10	299	90	0	0.70	112	0.99	0.99
T-RR-E-A-112	289	90	0	0.46	394	1.11	1.12
T-RR-E-A-119	293	90	0	0.46	256	1.10	1.10
T-RR-E-A-126	398	90	0	0.58	142	1.06	1.03
T-RR-C-A-41	298	90	0	0.69	394	1.06	1.07
T-RR-C-A-45	343	90	0	0.69	248	1.14	1.16
T-RR-C-A-49	290	90	0	0.69	163	1.14	1.15
Mean						1.07	1.08
CoV						0.054	0.059

**Table 15**Evaluation of proposed design methods, using the  $C_f$  factor, against test results of six RHS X joints classified as class b.

Specimen	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$2\gamma^*$	$2\gamma^*$ limit	$\eta^*$	$N_{1u}$ (kN)	$N_{1u}/N_{P,M}$	$N_{1u}/N_{P,LK}$
S1B1C11	409	90	0	20.8	37	0.50	316	1.26	1.29
S1B2C21	445	90	0	10.7	36	1.00	1207	1.04	1.07
S1B2C22	432	90	0	16.2	36	1.00	652	1.09	1.12
TF-100x50x4-100x50x4	952	90	0	12.8	24	1.00	494	1.39	1.47
X-1.0-32-700O	404	90	0	34.2	37	1.00	653	1.20	1.18
X-1.0-21-550O	418	90	0	22.9	37	1.00	1264	0.98	1.01
Mean								1.16	1.19
CoV								0.130	0.141

**Table 16**

Collated numerical results totalling eight RHS-to-RHS X joints under brace in-plane bending, reported by Yu [19].

Specimen	$b_0$ (mm)	$h_0$ (mm)	$t_0$ (mm)	$b_1$ (mm)	$h_1$ (mm)	$t_1$ (mm)	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$2\gamma^*$	$\eta^*$	$M_{1u,ip}$ (kNm)	$M_{1u,ip}/M_{y,ip}$
x10ie05	150	150	10.00	150	75	10.00	355	90	0	15	0.5	37.1	1.34
x10ie	150	150	10.00	150	150	10.00	355	90	0	15	1.0	89.7	1.26
x10ie2	150	150	10.00	150	300	10.00	355	90	0	15	2.0	259.7	1.19
x11i	150	150	6.25	150	150	6.25	355	90	0	24	1.0	50.0	1.37
x11ie2	150	150	6.25	150	300	6.25	355	90	0	24	2.0	128.7	1.06
x12ie05	150	150	4.29	150	75	4.29	355	90	0	35	0.5	12.2	1.72
x12i	150	150	4.29	150	150	4.29	355	90	0	35	1.0	28.6	1.28
x12ie2	150	150	4.29	150	300	4.29	355	90	0	35	2.0	76.5	0.97

**Table 17**

Evaluation of design methods for the loading case of brace in-plane bending.

Specimen	$f_{y0}$ (MPa)	$2\gamma^*$	$\eta^*$	$M_{1u,ip}$ (kNm)	$M_{1u,ip}/M_{C,M,ip}$	$M_{1u,ip}/M_{Yu,ip}$	$M_{1u,ip}/M_{Kuhn,ip}$	$M_{1u,ip}/M_{Lan,ip}$	$M_{1u,ip}/M_{P,M,ip}$	$M_{1u,ip}/M_{P,LK,ip}$
x10ie05	355	15	0.5	37.1	1.40	1.36	1.40	1.34	1.34	1.34
x10ie	355	15	1.0	89.7	1.33	1.25	1.32	1.26	1.33	1.34
x10ie2	355	15	2.0	259.7	1.25	1.24	1.25	1.19	1.39	1.41
x11i	355	24	1.0	50.0	1.63	1.24	1.64	1.37	1.63	1.65
x11ie2	355	24	2.0	128.7	1.25	1.18	1.26	1.15	1.39	1.41
x12ie05	355	35	0.5	12.2	2.48	1.38	2.48	1.84	2.23	2.22
x12i	355	35	1.0	28.6	1.84	1.11	1.84	1.54	1.84	1.82
x12ie2	355	35	2.0	76.5	1.40	1.29	1.40	1.40	1.55	1.54
Mean					1.57	1.25	1.57	1.39	1.59	1.59
CoV					0.265	0.072	0.265	0.159	0.198	0.191

Note: The material factor ( $C_T$ ) equals 1.0 for all the joints.

**Table 18**

Effects of chord sidewall convexity and thickness tolerance on the resistance of RHS-to-RHS T joints under brace in-plane bending (Nagui [42])

Specimen	$2\gamma$	Steel grade	$M_{1u,ip}$ (kNm)	$M_{C,ip}$ (kNm)	$M_{1u,ip}/M_{C,ip}$
t12i (convexity: 1%)	35	S355	28.9	22.9	1.26
	35	S460	34.5	29.7	1.16
	35	S700	50.3	46.1	1.09
t12i* (convexity: 1% + thickness : -10%)	35	S355	24.7	22.9	1.08
	35	S460	29.7	29.7	1.00
	35	S700	43.2	46.1	0.94
t13i (convexity: 1%)	30	S355	37.2	27.2	1.37
	30	S460	44.7	36.4	1.22
	30	S700	64.8	55.4	1.17
t13i* (convexity: 1% + thickness : -10%)	30	S355	31.9	27.2	1.17
	30	S460	38.4	36.4	1.05
	30	S700	55.8	55.4	1.01
Mean					1.13
CoV					0.109

**Table 19**

Collated numerical results totalling eight RHS-to-RHS X joints under brace out-of-plane bending, reported by Yu [19].

Specimen	$b_0$ (mm)	$h_0$ (mm)	$t_0$ (mm)	$b_1$ (mm)	$h_1$ (mm)	$t_1$ (mm)	$f_{y0}$ (MPa)	$\theta_1$ (°)	$n$	$2\gamma^*$	$\eta^*$	$M_{1u,op}$ (kNm)	$M_{1u,op}/M_{y,op}$
x10oe05	150	150	10.00	150	75	10.00	355	90	0	15	0.5	80.4	1.29
x10oe	150	150	10.00	150	150	10.00	355	90	0	15	1.0	119.4	1.20
x10oe2	150	150	10.00	150	300	10.00	355	90	0	15	2.0	192.5	1.11
x11o	150	150	6.25	150	150	6.25	355	90	0	24	1.0	59.4	1.03
x11oe2	150	150	6.25	150	300	6.25	355	90	0	24	2.0	108.6	1.03
x12oe05	150	150	4.29	150	75	4.29	355	90	0	35	0.5	23.0	1.08
x12o	150	150	4.29	150	150	4.29	355	90	0	35	1.0	37.2	0.98
x12oe2	150	150	4.29	150	300	4.29	355	90	0	35	2.0	62.7	0.88

**Table 20**

Evaluation of design methods for the loading case of brace out-of-plane bending.

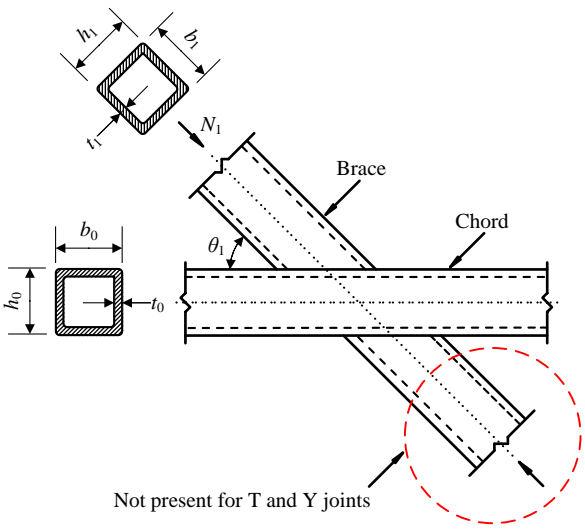
Specimen	$f_{y0}$ (MPa)	$2\gamma^*$	$\eta^*$	$M_{1u,op}$ (kNm)	$M_{1u,op}/M_{C,M,op}$	$M_{1u,op}/M_{Yu,op}$	$M_{1u,op}/M_{Kuhn,op}$	$M_{1u,op}/M_{Lan,op}$	$M_{1u,op}/M_{P,M,op}$	$M_{1u,op}/M_{P,LK,op}$
x10oe05	355	15	0.5	80.4	1.36	1.21	1.36	1.29	1.29	1.29
x10oe	355	15	1.0	119.4	1.26	1.14	1.26	1.20	1.26	1.28
x10oe2	355	15	2.0	192.5	1.16	1.06	1.16	1.11	1.29	1.31
x11o	355	24	1.0	59.4	1.22	1.09	1.23	1.03	1.22	1.23
x11oe2	355	24	2.0	108.6	1.22	1.12	1.23	1.12	1.35	1.37
x12oe05	355	35	0.5	23.0	1.55	1.30	1.56	1.15	1.39	1.39
x12o	355	35	1.0	37.2	1.41	1.26	1.41	1.18	1.41	1.40
x12oe2	355	35	2.0	62.7	1.26	1.17	1.27	1.25	1.40	1.39
Mean					1.30	1.17	1.31	1.17	1.33	1.33
CoV					0.097	0.070	0.097	0.073	0.054	0.046

Note: The material factor ( $C_T$ ) equals 1.0 for all the joints.



**Table 21**

Recommended design resistance for chord sidewall failure in RHS joints under brace axial compression using the modified bearing-buckling method.

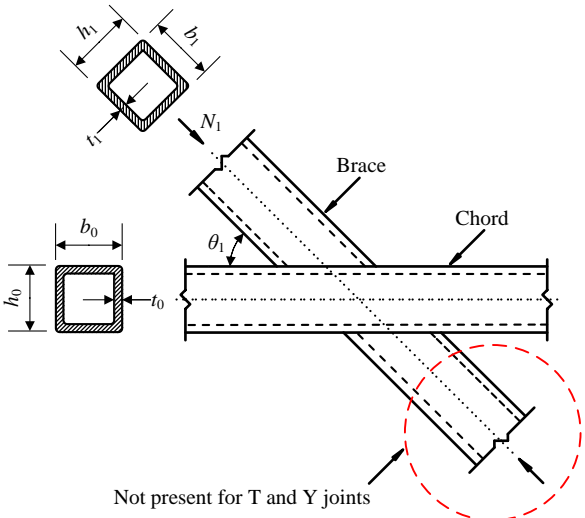
RHS-to-RHS X, T and Y joints (class a)	Brace axial compression loading	
 <p>Not present for T and Y joints</p>	$N_{1,Rd} = C_f f_k t_0 (2h_1 + 10t_0) \sqrt{\frac{1}{\sin \theta_1}} Q_f$	$f_k = \chi_{0.5} \left( \frac{h_0}{h_1} \right)^{0.15} \quad f_{y0} \leq f_{y0}$
	<b>Parameters</b>	
	$C_f = 1.1 - 0.1 f_{y0} / 355 \leq 1.0$	
	$\chi_{0.5}$ is the reduction factor for column buckling according to e.g., EN 1993-1-1 [6] using the buckling curve c, or an equivalent code/standard, and a normalised slenderness: $\lambda_{0.5} = \frac{1.73 \left( \frac{h_0}{t_0} - 2 \right)}{\pi \sqrt{\frac{E}{f_{y0}}}}$	
	$Q_f = (1 -  n )^{0.1}$ with $n$ in the connecting chord face $n = \frac{N_{0,Ed}}{N_{pl,0,Rd}} + \frac{M_{0,Ed}}{M_{pl,0,Rd}}$ for class 1 or 2 chord cross-sections under chord compression stress and for chord cross-sections under chord tension stress; $M_{el,Rd}$ should be used for class 3 chord cross-sections.	
<b>Validity ranges</b>		
steel grades up to S960; $\beta=1.0$ ; $b_0/t_0 \leq 40$ , $h_0/t_0 \leq 40$ ; $0.25 \leq h_1/h_0 \leq 2.0$ ; $0.5 \leq h_0/b_0 \leq 2.0$ ; $\theta_1 \geq 30^\circ$ .		

These recommendations may also be used for other X joints under brace axial compression:

- (1) Plate-to-RHS X joints (use  $t_1=h_1$ ) and RHS-to-RHS X joints both with nominal  $f_{y0} \leq 460$  MPa,  $\beta=1.0$  and  $h_1/h_0 \leq 0.25$ , but use  $f_k=f_{y0}$ .
- (2) For X joints (class a) with an RHS brace welded to one side of the chord and supported by a flat plate or another profile **welded** to the opposite side of the chord, the lower effective  $h_1$  on either side and  $\lambda_{0.5}$  should be used for the determination of the resistance.
- (3) For X joints (class b) with an RHS brace welded to one side of the chord and supported by a flat plate or another profile **unwelded** to the opposite side of the chord, the lower effective  $h_1$  on either side and  $\lambda_{0.7}=1.4\lambda_{0.5}$  should be used for the determination of the resistance. Here, the chord cross-section slenderness should not exceed class 3.
- (4) For other unspecified X joints (class c), the resistance should be determined using  $\lambda_{1.0}=2\lambda_{0.5}$ .

The cross-section slenderness of class 1, 2 and 3 is defined in national standards.

**Table 22**  
Recommended design resistance for chord sidewall failure in RHS joints under brace axial compression using the Lan-Kuhn method.

RHS-to-RHS X, T and Y joints (class a)	Brace axial compression loading	
 <p>Not present for T and Y joints</p>	$N_{1,Rd} = C_f f_k t_0 (2h_1 + 10t_0) \sqrt{\frac{1}{\sin \theta_1}} Q_f$	$f_k = \chi_{0.5} \left( \frac{h_0}{h_1} \right)^{0.15} \quad f_{y0} \leq f_{y0}$
	<b>Parameters</b>	
	$C_f = 1.1 - 0.1 f_{y0} / 355 \leq 1.0$	
	$\chi_{0.5}$ is the buckling reduction factor obtained from: $\chi_{0.5} = 1.12 - 0.012 \frac{h_0}{t_0} \sqrt{\frac{f_{y0}}{355}}$	
	$Q_f = (1 -  n )^{0.1}$ with $n$ in connecting chord face $n = \frac{N_{0,Ed}}{N_{pl,0,Rd}} + \frac{M_{0,Ed}}{M_{pl,0,Rd}}$ for class 1 or 2 chord cross-sections under chord compression stress and for chord cross-sections under chord tension stress; $M_{el,Rd}$ should be used for class 3 chord cross-sections.	
<b>Validity ranges</b>		
steel grades up to S960; $\beta=1.0$ ; $b_0/t_0 \leq 40$ , $h_0/t_0 \leq 40$ ; $0.25 \leq h_1/h_0 \leq 2.0$ ; $0.5 \leq h_0/b_0 \leq 2.0$ ; $\theta_1 \geq 30^\circ$ .		

These recommendations may also be used for other X joints under brace axial compression:

- (1) Plate-to-RHS X joints (use  $t_1=h_1$ ) and RHS-to-RHS X joints both with nominal  $f_{y0} \leq 460$  MPa,  $\beta=1.0$  and  $h_1/h_0 \leq 0.25$ , but use  $f_k=f_{y0}$ .
- (2) For X joints (class a) with an RHS brace welded to one side of the chord and supported by a flat plate or another profile **welded** to the opposite side of the chord, the lower effective  $h_1$  on either side and  $\chi_{0.5}$  should be used for the determination of the resistance.
- (3) For X joints (class b) with an RHS brace welded to one side of the chord and supported by a flat plate or another profile **unwelded** to the opposite side of the chord, the lower effective  $h_1$  on either side and  $\chi_{0.7}$  should be used for the determination of the resistance. The  $\chi_{0.7}$  value could be obtained from:  

$$\chi_{0.7} = 1.12 - 0.017 \frac{h_0}{t_0} \sqrt{\frac{f_{y0}}{355}}$$
 Here, the chord cross-section slenderness should not exceed class 3.
- (4) For other unspecified X joints (class c), the resistance should be determined using  $\lambda_{1,0}=2\lambda_{0,5}$ , see Table 21.

The cross-section slenderness of class 1, 2 and 3 is defined in national standards.